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20. ABSTRACT (Continue on reverse side if necessary and identity by block number)

A CLR single cylinder diesel engine is used to determine the effect of air content in the emulsified fuel on the performance, exhaust emissions and ignition delay of diesel engines.

The experiments were conducted using diesel fuel #2 and JP-4 as baseline fuels and emulsions containing 15%, 30% and 45% water by volume. The air charge temperature was varied from 88°F to 302°F. The effects associated with use of

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the emulsions on performance, ignition delay and exhaust emissions were determined by making detailed measurements of fuel consumption, engine operating parameters and exhause emissions.

The results showed that slight improvement in BSFC was observed when emulsions with low water content were used at high speeds. For high water content emulsions, the BSFC increased. Heating the intake air increased the BSFC at high engine loads and slightly improved it at low engine loads. While the ignition delay was found to increase with the increase of the water percentage in the emulsion, preheating the air charge was effective in reducing it. Although, NO, and soot formation were reduced effectively withthe increase of water content in the emulsion, preheating the air charge adversely affected NO_X and soot emissions. While CO and VHC emissions increased with the increase of the water content in the emulsion, increasing the intake air temperature slightly reduced ACO and VHC at low loads.

(n. Louis hydrocarbons)

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LIST OF SYMOBLS AND ABBREVIATIONS

ABDC after bottom dead cemter

A/D analog to digital converter

AF air fuel ratio

AMBT ambient air temperature
ATDC after top dead center

BHP brake horsepower

BTDC before top dead center

BMEP brake mean effective pressure

BP barometric pressure

BSFC brake specific fuel consumption

C constant

C.BHP corrected brake horsepower

CA crank angle

CFM cubic feet per minute

CFT temperature correction factor
CP(I) cylinder pressure amplitude

CYLVOL cylinder volume

D day

DENAIR density of air

DENDF density of diesel fuel
DENSUR density of surfactant

DENW density of water
DI direct injection

EM percent water in emulsion

E.TEMP exhaust gas temperature

EQMF equivalent mass flow rate of fuel

EVC exhaust valve opens
EVC exhaust valve closes

F force on dynamometer load cell

FC mass of fuel consumed

HLB hydrophile lipophile balance
HVDF heating value of diesel fuel
HVSUR heating value of surfactant
I number of tests per data set

ID ignition delay period
INJ fuel injection timing
IVC intake valve closes

IVO intake valve opens

LBF pounds force
LBM pounds mass

LEVER dynamometer lever arm length

M month

MA mass flow rate of air

MDF mass flow rate of diesel fuel
MF mass flow rate ot total fuel

MTH theoretical mass of air that fills the engine cylinder

each cycle at atmospheric conditions

MSUR mass flow rate of surfactant

N engine speed

NL(I) needle lift amplitude NUM number of data sets

OP exhaust smoke opacity

PCF pressure correction factor for HP calculation

PD pressure dorp across the meriam laminar flow element

PSI pounds per square inch

RAIR ideal gas constant for air

RFG rotational function generator

SCP(I) slope of the cylinder pressure curve

SNL(I) slope of the needle lift curve

SMNO value of smoke opacity monitor output (microamperes)

T time to consume 100 grams of fuel

TCF temperature correction factor for horsepower calculation

TEMPI intake air temperature
V.EFF volumetric efficiency

X volume flow rate of diesel fuel
XS volume flow rate of surfactant

Y total fuel flow rate

YE year

Z composite fuel density

LIST OF PUBLICATIONS

Portions of this study is now in preparation for publication in "Combustion and Flame".

LIST OF PARTICIPATING SCIENTIFIC PERSONNEL

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CHAPTER I

STATEMENT OF THE PROBLEM

Introduction

Experimental studies by several investigators [1-12] on the potential use of water/fuel emulsions as diesel engine fuel showed promising results.

Valdmanis and Wulfhorst [1] working on a single cylinder D.I. diesel engine found significant reductions in $\mathrm{NO}_{\mathbf{x}}$ and smoke emissions when water/ fuel emulsions were used. They also indicated that as the concentration of water in the emulsion was increased, the ignition delay increased. To compensate for that, they advanced the ignition timing which adversely affected the $NO_{\mathbf{X}}$ emission. Coon and Storment [2] working on a single cylinder CLR engine and using unstabilized water/fuel emulsions with water content varying from 2 percent to 23 percent by volume, reported a significant reduction in $\mathrm{NO}_{\mathbf{X}}$ emissions and exhaust smoke density. Increases in unburned HC and CO concentration were also encountered depending on speed, load and water content of the fuel. They also performed limited analysis of the effect of emulsified fuel on the ignition delay and best power timing of the engine. Their results did not show drastic changes in ignition delay with the increase of water content of the fuel. This may be attributed to the low percentage of water content used as compared to that used by Valdmanis and Wulfhorst [1]. Cook and Law [3] studied the effect of water/fuel emulsion on IMEP and particulate emission of a single cylinder diesel engine. Their results showed that with equal amount of fuel injection the IMEP was minimally affected with water addition. They also indicated that smoke emission was reduced when water content in the emulsion ranged between 10 percent and 20 percent by volume. However, excessive water addition was found to

outweigh this beneficial effect. They attributed this to the prolonged ignition delay at higher levels of water content in the emulsion. Crookes, et al. [4] studied the combustion of water/diesel fuel emulsions. Their results using a single cylinder 4-stroke diesel engine showed significant reduction in NO_X emission. A reduction in CO, unburned HC and soot concentrations were also obtained with increase in water content of the fuel up to a critical value.

In summary, most of the work performed on the use of water/fuel emulsions in diesel engines, to a great extent, showed improvements in $\mathrm{NO}_{\mathbf{x}}$ and soot emission characteristics of the engine and indicated also that an increase in the water content in the emulsion prolonged the ignition delay which can adversely affect engine performance. It is the purpose of this investigation to study the use of water/oil emulsions in diesel engines and their effect on ignition delay, engine performance and exhaust emissions.

Background

Research work on the ignition delay of neat hydrocarbon fuels has been carried by many investigators. Henein [13] gave an excellent review on the ignition delay in diesel engines. Henein and Bolt [14] studied the effect of fuel/air ratio, injection pressure, coolant temperature and turbulence on the ignition delay in diesel engines. They reported that increase in cylinder air pressure, fuel/air ratio, cooling water temperature and engine speed shortened the delay period. They also showed later [15] that increasing the air charge temperature was very effective in reducing the ignition delay for several fuels including gasoline. Hiroyasu [16] studied

the autoignition of fuel sprays in a constant volume bomb. He found that the ignition delay was primarily affected by the air temperature, pressure and oxygen concentration, all of which affect the rate of chemical reactions.

Gelalles [17], Gerrish and Ayers [18], and Pediani [19] studied the effect of preheating neat diesel fuels on the combustion and soot formation in diesel engines. Their results indicated that, in general, an increase in fuel temperature reduced the ignition delay and soot emissions, provided smoother combustion and finer atomization of spray but with reduced penetration.

Recent studies on the preheating of the fuel [20-22] showed that only fuel temperatures near their critical temperatures were effective in reducing the ignition delay. Unfortunately, at such temperature levels water separation and emulsion instabilities were found to occur [7].

In conclusion, it is noted that the major parameters which affect the ignition delay in diesel engines using neat hydrocarbon fuels are the air charge temperature, the preinjection fuel temperature, type of fuel, load and speed. It is the objective of this investigation to study parametrically the effect of above parameters on the ignition delay in diesel engines using water/fuel emulsions.

Objectives |

It is the objective of this investigation to study parametrically the effect of the following:

- l. The air charge temperature
- 2. The preinjection emulsified fuel temperature
- 3. Engine speed and load

- 4. Percentage of water content in the emulsified fuel
- 5. Type of fuel

on the ignition delay, performance and emissions in diesel engines.

CHAPTER II

THE EXPERIMENTAL APPARATUS

A CLR direct injection research diesel engine manufactured by Laboratory Equipment Company is used for this investigation. A complete list of engine specifications is given in Appendix A. The engine is coupled to an Eaton Dynamatic eddy—current dynamometer equipped with controls to regulate the engine speed and load.

The top end of the engine was completely rebuilt before engine tests were begun. The cylinder liner, piston rings, and head gasket, were replaced with new ones. The valve seats were also reconditioned and after the engine was reassembled the valve timing was checked to make sure it met new engine specifications.

Engine Systems and Instrumentation

Figures 1-4 show schematics of the engine systems and instrumentation.

Appendices B and C lists the range and precision of each instrument.

Intake Air System and Instrumentation

The intake air system consists of an air filter, an air flow meter, and a surge tank which contains heating elements to heat the intake air.

The surge tank and intake air plumbing leading to the engine are insulated with a 1/2 inch layer of Smooth Kote insulation.

Intake Air Flow Meter

The intake air flow is measured using a Meriam Laminar Flow Element.

The air flow meter is connected to the intake system before the surge tank

as shown in Figure 1. An automotive type air filter is connected to the other end of the flow meter to filter the incoming air.

The pressure drop across the flow meter is measured using an Ashcroft bourdon tube type pressure gauge.

Intake Air Heater and Controls

The intake air heater shown in Figure 1 is constructed by mounting four 1500-watt oven-type heating elements in the intake air surge tank. An Omega temperature Controller (model 1622) is used to regulate the intake air temperature by controlling the amount of power delivered to the heating elements. The temperature controller receives information from a thermocouple mounted in the intake air manifold.

Fuel Flow Measurement System

Figure 2 shows the fuel flow measurement system. A Fisher Ainsworth LC-1000 Digital Balance is used to measure the mass of fuel consumed. A Standard electric stop clock activated by a toggle switch measures the time required to consume a specified mass of fuel.

Combustion Instrumentation

Figures 3 and 5 show a schematic and a photograph of the combustion instrumentation while Figure 6 shows a typical oscilloscope display of the cylinder pressure, needle lift, fuel pressure, and degrees crank angle signals. The following transducers and signal processers/analyzers are used:

Cylinder Pressure Transducer. A Kistler piezoelectric pressure transducer (model 6121) is flush-mounted in the cylinder head. A Kiag-Swiss type

5002 dual charge amplifier is used to convert the charge from the piezoelectric transducer into a voltage signal. This voltage signal is connected to channel four of the oscilloscope and channel one of the Labmaster A/D converter.

Needle Lift Transducer. A Bently-Nevada proximity probe (type 190) is mounted on top of the fuel injector. The probe is connected to a model 3000 proximiter which uses a separate 18-volt power supply and outputs a voltage signal proportional to the displacement of the fuel injector needle. A DC offset amplifier is used to bring the needle lift voltage signal down to ground level. A Sanborn Differential Amplifier (model 8875A) amplifies this signal before it is sent to channel two of the Labmaster A/D converter and channel three of the oscilloscope.

Fuel Pressure Transducer. A Kistler pressure transducer (model 603B) is mounted on the fuel line between the fuel pump and the fuel injector. A charge amplifier built into channel one of the oscilloscopes 3A74 Engine Analyzer Amplifier converts this charge signal into a voltage signal so it can be displayed on the oscilloscope.

Top Dead Center Transducer. A magnetic transducer is mounted near the crankshaft. A thin piece of metal attached to the crankshaft passes by the magnetic transducer when the piston is in the TDC position. The signal from the magnetic transducer is sent to channel two of the oscilloscope.

Degrees Crank Angle Transducer. A Tektronix Rotational Function Generator is connected to one end of the engine crankshaft as shown in Figure

3. The signal from this generator is sent to the oscilloscopes 2B67 Engine

Analyzer Time Base module. The RFG produces small voltage spikes every ten degrees CA, larger spikes every 60 seconds CA, and one large spike at TDC. The signal from the magnetic TDC transducer is aligned with the TDC spike from the RFG to provide a meaningful time base for the oscilloscope which displays the engine cycle in degrees CA.

Oscilloscope. A Tektronix Type 564B Storage Oscilloscope is employed to display the signals from the combustion instrumentation transducers. The scope uses a four-channel Tektronix 3A74 Engine Analyzer module to amplify the cylinder pressure, needle lift, fuel pressure, and TDC transducer signals. A Tektronix 2B67 Engine Analyzer Time Base is used to accept the RFG signal. Photographs of the oscilloscope traces are taken with a Tektronix model C-27 oscilloscope camera.

Analog to Digital Converter. A Labmaster analog to digital converter is used to digitize the cylinder pressure and needle lift analog signals. The sampling frequency using two channels is 12.5 KHZ. Five hundred data points are collected per sample per channel yielding a resolution of 0.08 milliseconds per data point. The digitized cylinder pressure and needle lift signals are shown in Figure 7.

Computer System. An IBM Personal Computer is used to store data files containing samples of the needle lift and cylinder pressure digitized signals.

Input parameters for the data acquisition software are selected from an Amdek Video-300 graphics monitor using an FTG Data Lite Pen. An Epsom graphics printer prints out plots of the cylinder pressure and needle lift signals versus time (see Figure 7).

Load Measurement Instrumentation

An Emory hydraulic load cell measures the force transmitted from the dynamometer lever arm. A Heise bourdon tube type pressure gauge measures this hydraulic pressure and indicates the force on the load cell in pounds.

Engine Speed Measurement Instrumentation

The engine speed is measured using a magnetic pickup connected to a Standard electronic tachometer. The magnetic pickup is mounted next to a toothed-gear driven by the engine crankshaft.

Temperature Measurement Instrumentation

Thermocouples are used to measure the inlet and outlet cylinder head coolant temperature, the oil sump temperature, and the intake air temperature before the air flow meter. An Emory pyrometer is used to select the output from these thermocouples and indicate the temperature in degrees Fahrenheit.

The exhaust gas temperature is measured using a type K kromel--alumel thermocouple inserted in the exhaust manifold. An Omega thermocouple thermometer (model 660) measures the output from this thermocouple and indicates the exhaust gas temperature in degrees Celsius.

The intake air temperature just before the air enters the engine is measured using an Omega iron--constantan thermocouple. This thermocouple is connected to the Omega temperature controller (model 1622) which regulates the intake air temperature.

The ambient wet and dry bulb temperatures are measured using Taylor mercury in glass thermometers.

Barometric Pressure Measurement Instrumentation

The barometric pressure is measured using a Fisher Scientific mercury barometer.

Exhaust Smoke Opacity Monitor

The smoke opacity monitor shown in Figure 8 is mounted on the exhaust pipe to measure the opacity of the exhaust gases. A Beam of light is projected through the exhaust gas and is detected by a photo cell on the other side of the exhaust pipe. The smoke opacity monitor is equipped with removable slides which keeps the exhaust gases from entering the monitor when it is not in use. Air purge inlets on each end of the monitor allow compressed air to flow over the glass lenses to keep exhaust soot from collecting on the lenses while opacity readings are being taken. The glass lenses are also retractable for easy cleaning.

A Simpson DC microammeter is used to measure the output current from the photo cell.

Exhaust Emission Instrumentation

The exhaust emissions from the test engine are measured using the following instrumentation:

- 1. ${
 m NO}_{
 m X}$ concentration is measured using a Scott chemiluminescence analyzer. (Model 125)
- 2. CO and CO_2 concentration is measured using Beckman and nondispersive infrared analyzers (model 315A, 315B).

 Unburned hydrocarbon concentration is measured with an FID Beckman total hydrocarbon analyzer (Model 402).

A schematic of the instrumentation and the exhaust emission flow diagram is shown in Figure ${\bf 1}$.

CHAPTER III

TEST PROCEDURE

Engine Test Procedure

All performance tests are conducted in accordance with SAE J1349, the engine power test code for spark ignition and diesel engines. Any deviations from the test code are noted in the text.

Two neat fuels (diesel fuel #2 and JP-4) provided by the U. S. Army
Fuels and Lubricant Research Laboratory are used in this investigation.
Emulsified fuels having 15, 30, and 45 percent water content are also prepared for testing. The tests are conducted at engine speeds of 1000, 1500, and 2000 RPM at five engine loads (6, 9, 12, 15 and 18 foot pounds) and at three separate intake air temperatures of ambient, 167 and 302 degrees
Fahrenheit. The ambient temperature varies from 79 to 88°F.

To begin each test session, the engine is started on neat fuel and warmed up at the first test speed. The warm up period ends when the oil sump temperature reaches equilibrium. If the intake air is heated for the first test the temperature controller is set to the desired preheat temperature. The first test fuel is then selected. If emulsions are used the new fuel is introduced into the fuel lines and another period of time is allowed for the engine to stabilize. The fuel injection timing and rack setting are adjusted iteratively until optimum performance is obtained at the first load setting. A data sheet is then recorded with the following information: engine speed, engine load, intake air preheat temperature, inlet and outlet coolant temperatures, oil sump temperature, exhaust gas temperature, wet and dry bulb room temperatures, barometric pressure, oil

pressure, the pressure drop across the laminar flow element, fuel injection timing, rack setting, smoke opacity monitor reading, exhaust emission readings, and the fuel consumption time. A photo is taken of the oscilloscope screen which simultaneously displays the cylinder pressure, injector needle lift, fuel pressure, and degree crank angle signals. Two ignition delay period measurements are taken and stored in the IBM Personal Computer. The load is then increased to the next test value and the injection timing is again adjusted to obtain maximum performance. After the engine stabilizes another line of the data sheet is recorded. After all five load settings are tested the intake air preheat temperature, fuel type, and engine speed are changed in that order. See Figure 9 for a flow chart of the entire engine test program.

The general procedure then is to operate the engine on neat fuel at 1000 RPM for the five different loads and three separate intake air temperatures. Each temperature increase is followed by the five load settings. This sequence is repeated for the emulsified fuels containing 15, 30, and 45 percent water. The entire process is then repeated for the 1500 and 2000 RPM engine speeds.

Specific engine tests are repeated so a statistical analysis can be conducted on the data. Engine tests are repeated at 1000 and 2000 RPM for the three highest test loads of 12, 15, and 18 foot pounds using an intake air temperature of 88 degrees Fahrenheit. Fuels containing 0, 15, and 30 percent water are compared at these conditions to investigate the effect of increasing the water content in the fuel on the BSFC. Engine tests using 45 percent water/oil emulsion are not repeated due to the severe engine knock encountered during previous engine tests.

Calibration Procedure for the Exhaust Smoke Opacity Monitor

The exhaust smoke opacity monitor is calibrated using six light filters with optical densities of 0.10, 0.30, 0.40, 0.70, 1.00, and 2.00. The light source is first turned on and the gain is adjusted until the micro-ammeter reads 50 micro-amps. Each of the six filters is then placed between the light source and the photoelectric cell. The micro-ammeter is read after each filter is used. An equation which relates the optical density to opacity is then employed to determine the opacity corresponding to each filter:

Optical Density =
$$log \frac{1}{1 - Opacity}$$
.

A calibration curve (given in Appendix D) is then plotted to show the relationship between the micro-ammeter reading and the exhaust smoke opacity. The method of least squares is used to determine the equation of the straight line passing through the data points.

Emulsified Fuel Mixing Procedure

The water/oil emulsions are prepared by mixing the neat fuel and a constant 2 percent surfactant with 15, 30, and 45 percent deionized distilled water by volume as shown in Table I. The volumes of neat fuel, water, and surfactant are measured using graduated cylinders. The properties of #2 diesel fuel and JP-4 fuel are given in Appendices E and F.

The neat fuel and surfactant are first mixed together for two minutes.

The water is then added and the emulsion is blended for another ten minutes.

All emulsions are prepared just before each engine test using emulsified fuels.

The surfactant is prepared by mixing Span 80 (Sorbitan Monooleate HLB = 4.3) and Tween 85 (Polyoxyethylene (20) Sorbitan Trioleate HLB = 11.0) in a 3:1 mass ratio. These surfactants are purchased from ICI Chemicals, Inc., Wilmington, Delaware.

TABLE I FUEL COMPOSITION FOR EMULSIONS

Emulsion Type	Diesel Fuel Volume	Water Volume	Surfactant
15%	830 ml	150 ml	20 ml
30%	680 ml	300 ml	20 ml
45%	530 ml	450 ml	20 ml

Procedure for Determining the Fuel Heat of Combustion

The heat of combustion for the surfactant mixture used in this investigation is determined using a Parr Oxygen Calorimeter. The heat of combustion for pure diesel fuel #2 is also determined using this same calorimeter so the two heats of combustion will have the same reference base. (The heat of combustion for pure #2 diesel fuel calculated using this calorimeter differs by only 0.5% from the published net heat of combustion found in Appendix E.) These calculated heats of combustion for the surfactant and diesel fuel #2 are used in the data reduction program.

CHAPTER IV

DATA ANALYSIS

Data Reduction

Performance parameters are calculated to convert the test data into a more useful form. Values for the engine speed, load exerted on the dynamometer lever arm, mass of fuel consumed, time to consume the fuel, intake and ambient air temperatures, percent water in the fuel, pressure drop across the air flow meter, temperature correction factor for the air flow meter, barometric pressure, smoke number, exhaust gas temperature, fuel injection timing, and ignition delay period are taken from the engine data sheet and read into a computer program. The program performs all calculations in accordance with SAE J1349 and outputs values for the load exerted on the load cell, air and fuel flow rates, air fuel ratio, corrected brake horsepower, brake mean effective pressure, brake specific fuel consumption, fuel injection timing, ignition delay period, volumetric efficiency, exhaust gas temperature, and the smoke opacity. A copy of the data reduction program can be found in Appendix G. The output of this program for 1000 and 2000 RPM is given in Table III. Details of the calculations are given in Appendix H.

CHAPTER V

SUMMARY OF THE MOST IMPORTANT RESULTS

Summary of the Results

The results of this investigations can be summarized as follows:

- 1. Slight improvement in BSFC is noticed when w/diesel fuel emulsions are used as fuel. Higher water content in the emulsion will cause an increase in BSFC, while smaller water content will not significantly improve the BSFC of the engine.

 Statistically established BSFC results at 2000 RPM and 88°F ambient temperature show that adding 15 and 30 percent water to diesel fuel decrease the BSFC by an average of 2.54 and 1.92 percent respectively. A miximum decrease of 3.57 percent is obtained using 15 percent w/o emulsion (see Table II and Figure 10).
- 2. At low engine speed, no improvement in BSFC is noticed when w/o emulsions are used as fuel. Engine test results show that adding 15 and 30 percent water to diesel fuel increases the BSFC by an average of 2.12 and 3.83 percent respectively at 1000 RPM and 88°F ambient temperature. A maximum increase of 4.35 percent is obtained when 30 percent w/o emulsion is used. (see Table II and Figure 11).
- 3. Heating the intake air reduces the volumetric efficiency. (see Figures 12 and 13).
- 4. Heating the intake air increases the BSFC at high engine loads and may improve the BSFC at low engine loads. (see figures 14 21).
- 5. The ignition delay period increases with the increase of the water content of the emulsion. (see Figure 22)
- 6. Heating the intake air reduces the ignition delay period when neat and emulsionfied fuels are used. (see Figures 23, 27).

- 7. The test results obtained using JP4 as baseline fuel followed the same trend as that obtained when diesel fuel #2 was used as baseline fuel. However at high water content in the emulsion (45%) problems of starting and severe knock were noticed during testing. (see Figures 28, 34).
- 8. Preliminary tests on the preheating of the emulsion fied fuel show that at high temperatures water separation and emulsion instabilities occur. Thus, while preheating neat fuels at high temperatures improve the combustion characteristics of the engine, the heating of emulsion fied fuels is not beneficial.
- 9. While increasing the water content in the emulsion reduces the opacity of the exhaust gases. Increasing the intake air temperature increases the opacity of the exhause gases. (see Figures 35, 36)
- 10. While increasing the water content in the emulsion reduces ${\rm NO_X}$ emissions of the engine, increasing the intake air temperature increases ${\rm NO_X}$ emissions. (see Figures 37-38)
- 11. While increasing the water content in the emulsion increases unburned Hc and Co emissions of the engine, increasing the intake temperature reduces UHC and Co emissions at low loads. (See Figures 39-44)

Recommendations

- Basic experimental studies on the cooperation and combustion of w/o emulsion sprays both in bombs and engines should be conducted to determine the role played by the microexplosion phenomenon in improving the combustion characteristics of diesel engines with emphasis on the study of the most effective parameters involved.
- 2. The effect of use of emulsionfied fuels in turbocharged or supercharged diesel engines should be studied. Turbocharged engines can use high intake air temperatures thus reducing the ignition delay without scarificing the volumetric efficiency of the engine.
- 3. The effect of use of emulsionfied fuel in adiabatic diesel engines should be studied. It is anticipated that better combustion characteristics (short ignition relay), partial cooling of the engine (reduced thermal stresses), and better soot and NO_{X} emissions may result from this study.
- 4. Research work to develop w/o emulsions that can stay stable for longer periods of time is needed for future use of the emulsions as fuel in diesel engines.

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APPENDIX A

TABLE A-1

DIESEL ENGINE SPECIFICATIONS

Cylinder Bore 3.8125 inches
Stroke 3.750 inches

Displacement 42.810 cubic inches

Type of Cooling Water Cooled

Combustion Chamber Direct Injection

Piston Type "Mexican Hat"

Number of Compression Rings 2

Number of Oil Control Rings 1

Compression Ratio 16.7:1

Lubricating Oil Castrol 20W-50

IVO 18 Degrees BTDC

IVC 58 Degrees ABDC

EVO 124 Degrees ATDC

EVC 20 Degrees ATDC

Maximum Lift for Both Valves 0.228 inches

The intake valve is equipped with a 60 degree shroud to enhance the swirl component of the intake air.

FUEL INJECTION SPECIFICATIONS

Fuel Injection Pump Bosch APElB

Fuel Injection Timing 0-40 Degrees BTDC, Manual

Fuel Injector Nozzle Sims NL141

APPENDIX A CONTINUED

Number of Nozzle Holes 4

Nozzle Hole Diameter 0.27 mm

Injector Crack Pressure 2600 Psi

APPENDIX B

TABLE B-1

ENGINE INPUT PARAMETERS

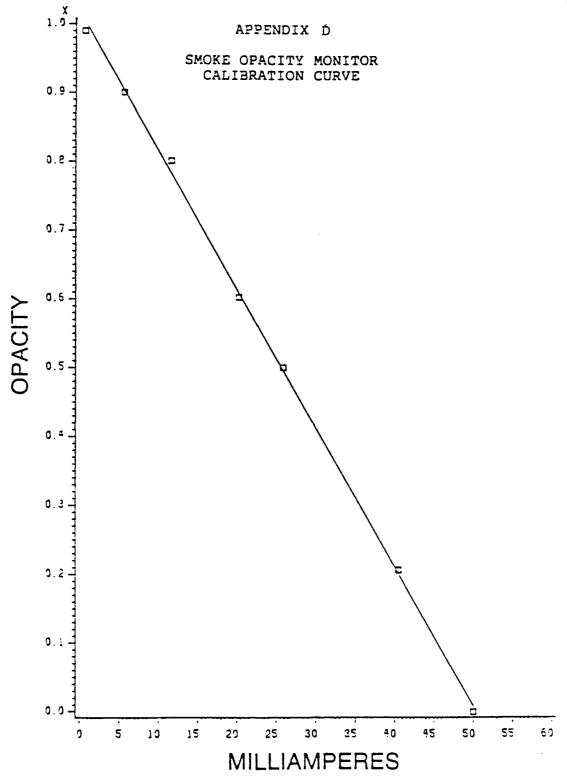
Parameter	Range	Precision (+ or -)	Instrumentation
Fuel Flow	1-1000 gr	0.01 gr	Fisher/Ainsworth LC-1000
Air Flow	0-200 CFM	0.5 CFM	Meriam LFE
Speed	0-7000 RPM	2.0 RPM	Magnetic Pickup
Fuel Injection Timing	0-50 BTDC	0.5 CA	Manual
Inlet Coolant Temperature	70-200 F	1.0 F	Thermo Electronic
Load	0-100 LBS	0.1 LB	BLE Load Cell
Atmospheric Pressure	20-32" Hg	0.01" Hg	Fisher Scientific Barometer
Intake Air Temp.	60-450 F	1.0 F	Omega Temp. Controller
Air Humidity	10-90 %	1.0 %	Wet And Dry Bulb Thermometer

APPENDIX C

TABLE C-1

ENGINE OUTPUT PARAMETERS

Parameter	Range	Precision (+ or -)	Instrumentation
Sm.Jke Opacity	0-100 %	1.0 %	Smoke Opacity Monitor
Exhaust Temp.	32-1400 F	1.0 F	Omega Model 660
Outlet Coolant Temperature	70-200 F	1.0 F	Thermo Electronic
Oil Gallery Temperature	70-400 F	1.0 F	Thermo Electronic
Fuel Injection Pressure	0-200 Bar	5.59 Pc/bar	Kistler (603b)
Cylinder Press.	0-250 Bar	14.0 Pc/bar	Kistler (6121)
Oil Press.	0-100 Psi	0.5 Psi	Press. Gauge
Needle Lift	0-1.0 mm	8.0 v/mm	Bently-Nevada (Probe Type 190, Proximiter Model 3000)



SMOKE OPACITY MONITOR CALIBRATION CURVE

APPENDIX E

TABLE E-1

PROPERTIES OF DIESEL FUEL #2

Gravity, API	34.6
Specific Gravity, g/ml	0.8519
Distillation, C	
IBP	199
10% Recovered	238
50% Recovered	272
90% Recovered	322
EP	355
Residue, Vol%	2.0
Loss, Vol%	0.0
Viscosity, cST at -20 Degrees C	Froze
Aromatics, FIA, Vol%	3.20
Olefins, FIA, Vol3	30.6
Mono-Aromatics, UV, wt%	6.85
Di-Aromatics, UV, wt%	7.82
Tri-Aromatics, UV, wt3	0.29
Hydrogen, wt%	12.95
Carbon, wt%	86.23
Nitrogen, wt%	0.01
Sulfur, wt%	0.380
Refactive Index at 20 Degrees C	1.4751
Carbon Residue, 10% Btms, wt%	0.15
Aniline Point, Degrees C (F)	74 (165)

APPENDIX E CONTINUED

Flash Point, Degrees C	74
Net Heat of Combustion Btu/lb	18,342
Calculated Net Heat of Combustion	18,543
Cetane Number	50.1
Cetane Index (D 976-80)	47.7

The following heats of combustion are determined in the lab using a Parr Oxygen Bomb Calorimeter. These two values are used in the data reduction program.

of Combustion (Btu/lb)	for	#2 diesel	18,449.14
of Combustion		the	14,123.96

PROPERTIES OF JP-4 FUEL

TABLE F-1

	AL-10583-T
Gravity, OAPI	54.0
Specific Gravity, g/mL	0.7628
Distillation, C	
IBP	<i>5</i> 8
10% Recovered	91
50% Recovered	149
90% Recovered	229
EP	255
Residue, vol%	1.5
Loss, vol%	0
Viscosity, cSt at -20°C	1.98
Viscosity, cSt at 40°C	0.78
Aromatics, FIA, vol%	14.8
Olefins, FIA, vol%	1.0
Mono-Aromatics, UV, wt%	10.60
Di-Aromatics, UV, wt%	1.19
Tri-Aromatics, UV, wt%	0.01
Hydrogen, wt%	14.34
Carbon, wt%	85.65
Nitrogen, wt%	< 0.01
Sulfur, wt%	0.018
Refactive Index at 20°C	1.4336
Carbon Residue, 10% Btms, wt%	0.07
Aniline Point, °C(°F)	57(135)
Flash Point, °C	-22
Net Heat of Combustion,	
MJ/kg	43.303
Btu/lb	18617
Calculated Net Heat of Combustion,	
MJ/kg	43.606
Btu/lb	18747
Cetane Number	34.5
Cetane Index (D 976-80)	34.7

APPENDIX G

DATA REDUCTION PROGRAM

```
10 FOR K=1 TO 30
       IF K=1 THEN 30 ELSE 170
LPRINT "UNITS ARE: "
20
JO LPRINT "UNITS ARE: "

40 LPRINT

50 LPRINT "LOAD (LD): LBF"

60 LPRINT "AIR FLOW RATE (AIR): LBM / MIN"

70 LPRINT "FUEL FLOW RATE (FUEL): LBM / MIN"

80 LPRINT "AIR FUEL RATIO (A/F): NONE"

90 LPRINT "GGRECTED BRAZE HORSEPOWER (C.BHP): HP"

100 LPRINT "BRAKE MEAN EFFECTIVE PRESSURE (BMEP): PSI"

110 LPRINT "INJECTION TIMING (INJ): DEG. CA BIDC"

120 LPRINT "IGNITION DELAY PERIOD (ID): MILLISECONDS"

130 LPRINT "VOLUMETRIC EFFICIENCY (V.EFF): Z"

140 LPRINT "EIHAUST GAS TEMPERATURE (E.TEMP): DEG. FARENHEIGHT"

150 LPRINT "EIHAUST SMORE OPACITY (OP): Z"
 30
 150 LPRINT
 170 READ NUM, I, M, D, YE, EM, TEMPI, N, AMBT
 180 FOR J=1 TO I
 190 READ F, PD, FG. T. BP, CFT, SMNO, E. TEMP, INJ, ID
200 IF J>1 THEN
210 LPRINT
220 LPRINT "----
                         THEN 300
                                                   ELSE 210
 230 LPRINT "ENGINE SPEED ="N "RPM"" PERC
240 LPRINT " INTAKE AIR TEMP.="TEMPI "F"
                                                                             PERCENT WATER ="EM;
 250 LPRINT
250 LFRINT "LD";TAB(6)"AIR";TAB(12)"FUEL";TAB(20)"A/F";TAB(27)"C.BHP";
270 LPRINT TAB(34)"BMEP";TAB(42)"BSFC";TAB(49)"INJ";TAB(56)"ID";TAB(61)"V.EFF";
280 LPRINT TAB(68)"E.TEMP";TAB(76)"ID"
 290 LPRINT
 300 DENDF - .9519
 310 DENSUR = .9996
 320 DENW = 1!
 330 HVSUR = 14123.9638#
 340 HVDF = 18449.145#
 350 RAIR = 53.331
 350 KALK = 33.331
360 CYLVOL = 42.809
370 LEVER = .75
380 IF EM=0! TREN 470 ELSE 390
390 Z = (EM=DENW-(100!-(EM+2!))=DENDF-2!=DENSUR)/100!
400 Y = FC/(Z=T)
 410 \text{ X} = \text{Y}^{\pm}(100! - (\text{EM} + 2))/100!
 420 XS = Y=2!/100!
430 MDF = X=DENDF=2.205/1000!
 440 MSUR = XS=DENSUR=2.205/1000!
 430 EQMF = MDF = 1! +MSUR = HVSUR/HVDF
 460 MF - EQMF
 470 CFM - PD=100!/8!
 480 DENAIR = BP*14.696*144/(RAIR*29.92*(AMBT+460))
 490 MA = CFM=BP*CFT*DENAIR/(29.92)
500 IF EM=0! THEN 510 ELSE 530
  510 \text{ MDF} = FC/(453.597*T)
  520 MF - MDF
  530 AF - MA/MF
  540 MTH - DENAIR*CYLVOL*N/(2*12*3!)
  550 V.EFF - MA=100!/MTH
 560 BHP = F*LEYER*2!*3.14159*N/33000!

570 PCF = (29.3139/BP)*1.1

580 TCF = ((AMBT+460!)/537!)*1.2
 590 C.BHP - BHP*(PCF*TCF)*.3
600 IF EM-0! THEN 610 ELSE 640
  610 BSFC - MDF - 601/C.BHP
 620 GOTO 640
 630 BSFC - EOMF * 50! / C. BHP
```

APPENDIX G CONTINUED

```
640 BMEP = BHPC*33000!*2!*12!/(CYLVOL*N)
650 IF SMNO<=1! THEN 660 ELSE 660
660 OP = 0!
670 GOTO 690
680 OP = 101.39-1.99*SMNO
690 LPRINT USING "##.";F;
700 LPRINT USING "###.###";AF;
710 LPRINT USING "###.###";AF;
720 LPRINT USING "###.###";C.BHP;
730 LPRINT USING "###.###";C.BHP;
730 LPRINT USING "###.###";BMEP;
740 LPRINT USING "###.###";BSFC;
750 LPRINT USING "###.##";ID;
750 LPRINT USING "###.##";ID;
760 LPRINT USING "####.##";E.TEMP;
770 LPRINT USING "####.##";E.TEMP;
780 LPRINT USING "####.##";E.TEMP;
790 LPRINT USING "###.##";OP
```

APPENDIX H

DETAILS OF CALCULATIONS

Brake Horsepower

The brake horsepower is determined from a product of the dynamometer lever arm length, the load placed on the arm, and the engine speed.

$$BHP = L \times F \times N \times C \tag{1}$$

Corrected Brake Horsepower

The corrected brake horsepower is determined using the following pressure and temperature correction factors:

$$PCF = \frac{29.3139}{BP} 1.1 \tag{2}$$

$$TCF = \frac{AMBT + 460}{537} \cdot 1.2$$
 (3)

The corrected brake horsepower is then calculated using the equation:

$$C.BHP = BHP (PCF \times TCF)^{0.3}$$
 (4)

Brake Mean Effective Pressure

The brake mean effective pressure is calculated by dividing the corrected brake horsepower by the engine displacement.

$$\frac{\text{BMEP} = \frac{\text{BHPC}}{\text{CYLVOL}}}{(5)}$$

Fuel Flow Rate

The fuel flow rate is determined by measuring the amount of time required for the engine to consume a specified mass of fuel. This flow rate is calculated by dividing the mass of fuel consumed by the time to consume it.

Mass Flow Rate of Neat Fuel

The mass flow rate of neat fuel is calculated using the equation:

$$MDF = \frac{FC}{T} \tag{6}$$

Mass Flow Rate of Emulsified Fuel

When emulsified fuels are used corrections for the change in fuel composition have to be made since the fuel contains diesel fuel, surfactant, and various percentages of water.

The densities of diesel fuel #2, water, and surfactant are multiplied by their volume percentages in the fuel and added together to determine a composite fuel density.

$$Z = [EM \times DENW + (100 - (EM + 2)) \times DENDF + 2 \times DENSUR] / 100$$
 (7)

The volume fuel flow rate is then found by dividing the mass of fuel consumed by the composite fuel density and the time required to consume the fuel.

$$Y = \frac{FC}{Z \times T} \tag{8}$$

Separate volume flow rates of diesel fuel and surfactant are determined by multiplying the total fuel flow rate by the volume percentages of the diesel fuel and surfactant respectively.

$$X = \frac{Y \times (100 - (EM + 2))}{100}$$
 (9)

$$XS = \underbrace{Y \times 2}_{100} \tag{10}$$

The mass flow rates of diesel fuel and surfactant are calculated by multiplying the volume flow rate of each component by their respective densities.

$$MDF = X \times DENDF \tag{11}$$

$$MSUR = XS \times DENSUR \tag{12}$$

An equivalent fuel mass flow rate is then determined because the heating value of the surfactant is 23.44 percent lower than the heating value of pure diesel fuel #2. This calculation is necessary so comparisons in fuel consumption can be made between engine tests using neat and emulsified fuels. The equivalent fuel mass flow rate is found by correcting the surfactant flow rate. The surfactant flow rate is multiplied by the ratio of the heating value of surfactant to the heating value of diesel fuel. Thus the equivalent mass flow rate of fuel when using emulsions is based on the heating value of pure diesel fuel and found by adding the corrected flow rate of surfactant to the flow rate of pure diesel fuel.

$$EQMF = MDF \times 1 + MSUR \underbrace{HVSUR}_{HVDF}$$
 (13)

Mass Air Flow Rate

The air flow rate is calculated by including the calibration curve for the Meriam laminar flow element in the data reduction program. The pressure drop across the element is read into the program and the calibration curve is used to determine the air flow rate in cubic feel per minute.

$$CFM = \frac{PD \times 100}{8}$$
 (14)

The air density is then calculated using the ideal gas law.

$$DENAIR = \frac{BP}{RAIR \times AMBT}$$
 (15)

The air flow is corrected for temperature and pressure variations using the correction factors supplied by the manufacturer of the Meriam laminar flow element.

$$MA = \frac{CFM \times BP \times CFT \times DENAIR}{29.92}$$
 (16)

Air Fuel Ratio

The air fuel ratio is calculated by dividing the air flow rate by the fuel flow rate.

Air Fuel Ratio Using Neat Fuel

The air fuel ratio is calculated for neat fuel using the equation:

$$AF = \frac{MA}{MDF} \tag{17}$$

Air Fuel Ratio Using Emulsified Fuels

The air fuel ratio is calculate for emulsified fuels using the equation:

$$AF = \frac{MA}{EOMF}$$
 (18)

Brake Specific Fuel Consumption

The brake specific fuel consumption is calculated by dividing the mass flow rate of fuel by the corrected brake horsepower.

BSFC Using Neat Fuel

The brake specific fuel consumption is calculated for neat fuel using the equation:

$$BSFC = \frac{MDF}{C.BHP}$$
 (19)

BSFC Using Emulsified Fuel

The brake specific fuel consumption is calculated for water/oil emulsions using the equation:

$$BSFC = \frac{EQMF}{C.BHP}$$
 (20)

Volumetric Efficiency

The volumetric efficiency is calculated by dividing the air flow rate per cycle by the theoretical mass of air that would fill the engine displacement at ambient temperature and pressure. The theoretical mass

$$MTH = \frac{DENAIR \times CYLVOL \times N}{2}$$
 (21)

The volumetric efficiency is then calculated using the equation:

$$VEFF = \frac{MA}{MTH}$$
 (22)

The Exhaust Smoke Opacity

The exhaust smoke opacity is determined using the equation for the smoke opacity monitor calibration curve found in Appendix I.

$$OP = 101.39 - 1.99 \times SMNO$$
 (23)

Ignition Delay Period

The ignition delay period is determined using two methods. The first method is to analyze a photograph of the cylinder pressure, fuel injector needle lift, and degrees crank angle signals taken from the oscilloscope. The points where the fuel injector opens and the pressure rises due to combustion are compared against the crank angle degrees markers to determine the ignition delay period with units of degrees crank angle. This method however, is not very accurate since the crank angle markers can only be resolved to plus or minus 5.0 degrees CA.

The second method employs a Labmaster analog to digital converter and an IBM personal computer to digitize and store the needle lift and cylinder pressure analog signals. The sampling frequency using two channels is 12.5 KHz. Five hundred data points for each channel are stored per sample yielding a resolution of 0.08 milliseconds per data point. Initially a program was written to calculate the igniton delay period by mathematically analyzing the two digitized curves. The program determined the beginning of the ignition delay period by noting when the injector needle moved a certain distance (displayed as a voltage amplitude) from its equilibrium closed position. The end of the ignition delay period was determined by differentiating the

cylinder pressure curve assuming that the pressure rise due to combustion coincided with the maximum pressure gradient. This program however, proved to be inacurate and inconsistent since the maximum pressure gradient in the cylinder did not always coincide with the beginning of combustion.

The ignition delay period is finally determined by printing out the digitized data points (voltage values) of the needle lift and cylinder pressure signals. See Table IV. These points are inspected visually to determine where the fuel injector opens and the cylinder pressure increases due to combustion. The number of data points between the opening of the fuel injector and the beginning of combustion are counted. A value for the pressure rise ignition delay period is then determined by multiplying this number of data points by the time (0.08 milliseconds) between the digitized voltage values.

ID = # of Data Points x 0.08 Milliseconds (24)

This method is labor intensive, but proves to be the most accurate. The ignition delay period can be determined to within plus or minus 0.08 milliseconds. The computer is used to store two sets of ignition delay data for each engine test condition. These two ignition delay period values are later averaged and recorded. The two methods of determining the ignition delay period as described above are compared and found to agree with each other. The second method however, is able to achieve higher resolution since 0.08 milliseconds (the time interval between the digitized data points when using a sampling frequency of 12.5 KHz) corresponds to .48 and .96 degrees crank angle at engine speeds of 1000 and 2000 RPM respectively.

Sample Calculations

The following calculations are performed for data sets 13 and 16 which correspond to engine tests using neat and emulsified fuels respectively. Calculations are performed for the 24 lb load in each case. The corresponding output of the data reduction program is given in Table III.

Input Parameters for Data Set 13

Percent Water in the Fuel (EM) = 0.0 % 79.0 F Intake Air Temperature (TEMPI) Engine Speed (N) 2000 RPM = 79.0 FAmbient Air Temperature (AMBT) Load on Dynomometer Arm (F) = 24 LB Pressure Drop Across the Air Flow Meter (PD) = 1.595 " H20 Mass of Fuel Consumed (FC) = 100.00 GMTime to Consume the Fuel (T) 3.592 MIN Barometric Pressure (BP) 29.810 " HG Temperature Correction Factor for the Air Flow Meter (CFT) 0.9674 Smoke Number (SMNO) 37.50 mA

Calculation of Performance Parameters for Data Set Number 13 (Neat Fuel)

From Equation (1):

Brake Horsepower =
$$\frac{0.75 \times 24.0 \times 2 \times 3.14159 \times 2000}{33000}$$

6.854 HP

From Equations (2) and (3):

Pressure Correction Factor

$$= \frac{29.3139}{29.810} \cdot 1.1$$

= 0.9817

Temperature Correction Factor

$$= \frac{79.0 + 460}{537} \cdot 1.2$$

= 1.0045

From Equation (4):

Corrected Brake Horsepower

$$= 6.854 (0.9817 \times 1.0045^{0.3})$$

= 6.826 HP

From Equation (5):

Brake Mean Effective Pressure

$$= \frac{6.826 \times 33000 \times 2 \times 12}{42.809 \times 2000}$$

= 63.140 PSI

From Equation (6):

Mass Flow Rate of Diesel Fuel

$$= \frac{100.00}{3.592 \times 453.597}$$
$$= 0.0614 LBM$$

From Equation (14):

Air Density =
$$\frac{29.810 \times 14.696 \times 144}{(53.331 \times 29.92 (79.0 + 460))}$$

$$= 0.0733 LBM / CC$$

Combining Equations (15) and (16):

Air Mass Flow Rate

From Equation (17):

Air Fuel Ratio =
$$\frac{1.4086}{0.0614}$$
 = 22.9414

From Equation (19):

Brake Specific Fuel Consumption

$$= \frac{0.0614 \times 60}{6.826}$$

= 0.5395 LBM FUEL / HP HR

From Equation (21):

Theoretical Mass of Air

=
$$\frac{0.0733 \times 42.809 \times 2000}{2 \times 12 \times 12 \times 12}$$

= 1.8159 LBM

From Equation (22):

Volumetric Efficiency

$$= \frac{1.4086 \times 100}{1.8159}$$
$$= 77.57 \%$$

From Equation (23):

Exhaust Smoke Opacity

Input Parameters for Data Set 16

Percent Water in the Fuel (EM)	=	15.0 %
Intake Air Temperature (TEMPI)	=	81.0 F
Engine Speed (N)	=	2000 RPM
Ambient Air Temperature (AMBT)	=	81.0 F
Loan on Dynomometer Arm (F)	=	24.0 LB
Pressure Drop Across the Air Flow Meter (PD)	=	1.655 " H20
Mass of Fuel Consumed (FC)	=	100.00 GM
Time to Consume the Fuel (T)	=	2.998 MIN
Barometric Pressure (BP)	=	29.60 " HG
Temperature Correction Factor for the Air		
Flow Meter (CFT)	=	0.9611
Smoke Number (SMNO)	=	45.0 mA

Calculation of Performance Parameters for Data Set Number 16 (Emulsified Fuel)

The equations used to calculate the performance parameters when using emulsified fuels are the same as the neat fuel equations except that the flow rate of diesel fuel is replaced with the flow rate of emulsified fuel. The calculations of the corrected brake horsepower, brake mean effective pressure, air flow rate, volumetric efficiency, and smoke opacity are identical to the calculations performed in section 4.2.2. Only the results needed for further calculations are shown here.

C.BHP = 6.851 HP BMEP = 63.37 PSI MA = 1.427 LBM / MIN

The mass flow rate of emulsified fuel (EQMF) is found using equations (7) through (13).

From Equation (7):

Composite Fuel Density

=
$$[15 \times 1.0 + (100 - (15 + 2)) \times 0.8519 + 2 \times 0.9996] / 100$$

= 0.8771 GM / CC

From Equation (8):

Total Volume Fuel Flow Rate

$$= \frac{100}{0.8771 \times 2.998}$$

= 38.0307 CC/ MIN

From Equation (9):

Volume Flow Rate of Diesel Fuel

$$= \frac{38.0307 \times (100 - (15 + 2))}{100}$$

= 31.5655 CC / MIN

From Equation (10):

Volume Rate of Surfactant

$$= \frac{38.0307 \times 2}{100}$$

= 0.7606 CC / MIN.

From Equation (11):

Mass Flow Rate of Diesel Fuel

$$= \underbrace{31.5655 \times 0.8519 \times 2.205}_{1000}$$

= .05929 LBM / MIN

From Equation (12):

Mass Flow Rate of Surfactant

$$= \frac{0.7606 \times 0.9996 \times 2.205}{1000}$$

= 0.00676 LBM / MIN

From Equation (13):

Equivalent Mass Flow Rate of Fuel

$$= \frac{0.0529 \times 1 + 0.00676 \times 14123.9658}{18449.145}$$

= 0.060577 LBM / MIN

From Equation (18):

Air Fuel Ratio =
$$\frac{1.427}{0.060557}$$

= 23.557

From Equation (20):

Brake Specific Fuel Consumption

$$= \frac{6.0577 \times 10 \times 60}{6.851}$$

= 0.5305 LBM FUEL / HP HR

Ignition Delay Period Calculation

Table IV is the output of the computer program which prints out the digitized voltage values for the cylinder pressure and needle lift signals. Figure—is the plot of they cylinder pressure and needle lift signals corresponding to Table IV. The data acquisition software is written such taht a 10 volt input to the A/D converter corresponds to an amplitude of 2047. The first column contains the data point number from 1 to 500 since 500 data points are collected per channel. Columns 2 - 4 contain the needle lift amplitude, slop of the needle lift curve, cylinder pressure amplitude, and slope of the cylinder pressure curve respectively. By inspecting the needle lift and cylinder pressure slopes it is evident that the fuel injector needle begins to open at data point 226 and the cylinder pressure increases due to combustion at data point 240. The ignition delay period is then calculated using equation 24.

From Equation (24):

Ignition Delay Period = $14 \times .08 \text{ MILLISECONDS}$

= 1.12 MILLISECONDS

TABLE II
STATISTICALLY ESTABLISHED BSFC RESULTS

Engine	BMEP	H20 In Fuel	Ave BSFC	Percent
RPM	<u>Psi</u>	Vol. %	Lbm/Bhp Hr	Change
1000	41.9	0.0	0.5864	
1.1	1 1	15.0	0.5960	+1.64
• •	1.1	30.0	0.6108	+4.16
• •	52.4	0.0	0.5516	
1 1	1.1	15.0	0.5673	+2.85
1.1	1.1	30.0	0.5756	÷4.35
1 1	62.9	0.0	0.5420	
	1.1	15.0	0.5520	+1.88
1.1	7. 1	30.0	. 0.5581	+2.97
2000	42.1	0.0	0.6124	
1.1	• •	15.0	.0.6014	-1.80
1 1	1.1	30.0	0.6024	-1.34
1.,	52.6	0.0	0.5709	
1.1	1 1	15.0	0.5580	-2.26
1.1	• •	30.0	0.5586	-2.15
1.1	63.1	0.0	0.5458	
• •	1.1	15.0	0.5263	-3.57
1 1	• •	30.0	0.5334	-2.27

TABLE III

SAMPLE OF COMPUTED RESULTS

UNITS ARE :

LOAD (LD): LBF
AIR FLOW RATE (AIR): LBM / MIN
FUEL FLOW RATE (FUEL): LBM / MIN
AIR FUEL RATIO (A/F): NONE
CORRECTED BRAKE HORSEPOWER (C.BHF): HP
BRAKE MEAN EFFECTIVE PRESSURE (BMEF): PSI
BRAKE SPECIFIC FUEL CONSUMPTION (BSFC): LBM FUEL / HP HR
INJECTION TIMING (INJ): DEG. CA BTDC
IGNITION DELAY PERIOD (ID): MILLISECONDS
VOLUMETRIC EFFICIENCY (V.EFF): %
EXHAUST GAS TEMPERATURE (E.TEMP): DEG. FARENHEIGHT
EXHAUST SMOKE OPACITY (OP): %

===	======	======		1	22 25 25		3257E2		* ####		
ENG	INE SPE	ED = 10	QO RPM	PERC	ENT WAT	ER = 0	IN	TAKE A	IR TEMP	.= 81.5	F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
8	0.652	0.015	44.112	1.142	21.13	0.7759	23.0	1.44	72.56	431.6	1.89
12	0.647	0.018	35.955	1.713	31.70	0.6304	22.5	1.36	72.08	537.8	1.89
16	0.646	0.022	29,620	2.284	42.26	0.5730	21.0	1.28	71.96	633.2	1.89
20	0.642	0.026	24.988	2.855	52.83	0.5398	22.0	1.20	71.48	725.0	8.26
24	0.641	0.030	21.453	3.427	63.39	0.5231			71.36	815.0	17.81
===		=======		2	===		=====	RBESSE	=====		
			OO RPM							.= 167 F	•
LD	AIR	FUEL	A/F	C.BHF	BMEP	BSFC	INJ	ID	V.EFF	E. TEMP	% OF
8	0.619	0.015	40.862	1.143	21.14	0.7956	23.0	1.76	69.00	539.€	1.89
12	0.607	0.019	31.862	1.714		0.6671			67.70		
16	0.606	0.023	26.469				27.0		67.49		
20	0.596	0.027							66.41		
24		0.031			-				65.45		
47	Q.J6/	0.001	10.0/-	2.72/	00.40	0.0000	20.0	1.1-	63.75	000.0	_0.10
===	======	======	*****		===	222222	=====	Z# = # # #			
			00 RPM							.= 302 F	.
LD	AIR	FUEL	A/F	C.BHF	EMEF	BSFC	INJ	ID	V.EFF	E.TEMP	% OF
3	0.571	0.016	35.109	1.178	21.06	0.8571	22.0	0.96	60.00	638.6	7.26
12	0.561	0.020	28.126	1.708	51.59	0.7011	22.0	1.04	61.92		
	0.551	0.024		2.277	42.12	0.6417	21.0	0.92	60.77	851.0	14.23
	0.546	0.027	18.681	2.846	52.65	0.6159	26.0	1.00	60.18	923.0	26.17
24	0.540	0.037	14.673	3.415	63.18	0.6460	28.5	0.96	59.50	1041.8	45.47
			=======								
ENG	INE SPE	ED = 10	000 RPM	PERC	ENT WAT	ER = 15	I	NTAKE	AIR TEM	IF.= 79.5	5 F
LD	AIR	FUEL	A/F	C.BHF	BMEF	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
8	0.665	0.016	41.771	1.178	21.05	0.8399	22.0	1.48	73.30	456.8	1.89
12	0.643	0.019		1.707		0.6846	21.0	1.44	70.84	537.8	1.89
16	9.678	0.023		2.276		0.6144		1.48	70.28	638.6	1.89
20	0.544					0.5740		1.44	70.93	725.0	5.08
24	0.640	0.071		3.413	63.15	0.5467		1.44	70.45	824.0	8.85
	7.0 7 0	17 a 17 a 4	_ · · · · · / ·	w. 	9						

			OO RPM			ER = 15				IP.= 167	F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
8	0.622	0.016	39.015	1.141	21.10				68.9 3	527.0	
	0.612	0.019		1.711	31.65		20.5		67.83		
16 20	0.611	0.023 0.027	26.281 22.741	2.281 2.852	42.21 52.76	0.6111 0.5587	21.0	1.36		705.2 807.8	
24	0.603	0.031	19.421	3.423			23.0	1.28		892.4	
===			======	6	===	******	=====	=====	======		
ENG	INE SPE	ED = 10	00 RPM	PERC	ENT WAT	ER = 15	I	NTAKE	AIR TEM	MP.= 302	F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ΙD	V.EFF	E.TEMP	% OP
8	0.565	0.014	39.249	1.141		0.7575		1.24	62.68		
	0.561	0.019	28.981	1.711			21.0	1.24			
16		0.024	23.284	2.281	42.21		22.0			802.4	
20 24	0.551 0.541	0.029 0.034	19.193 15.920	2.852	52.76 63.32		22.5 24.0		61.14 60.05		
	0.541	0.054	13.720	J. 422	00.02	0.5761	27.0	1.04	80.03	1011.2	20.17
										rc - 00 c	
ENG	INE SPE	.ED = 10	00 RPM	PERC	ENT WAT	FK = 30	1	NIAKE	AIR IEM	F.= 82 F	•
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
_	0.658	0.016		1.140					72.76	420.8	1.89
		0.019		1.709			23.0		72.64		
16	0.651	0.023	28.829	2.279	42.17		23.0		72.04		1.89
20	0.646 0.636	0.027 0.030	24.268	2.849	52.71		23.0 23.0	1.84	71.44 70.35	692.6 784.4	5.47 6.86
24	0.000	0.050	21.056	3.419	63.25	0.0002	25.0	1.76	70.33		0.00
			**=====	_							
			000 RPM	_						IP.= 167	F
ENG			000 RPM	_		ER = 30		NTAKE	AIR TEM	IP.= 167 E.TEMP	
ENG LD	INE SPE	ED = 10	A/F 39.252	PERC	BMEP	ER = 30 BSFC 0.8286	I INJ 24.0	NTAKE ID 1.72	AIR TEM		% OF
ENG LD 8 12	O.618	FUEL 0.016 0.019	A/F 39.252 32.329	PERC C.BHF 1.140 1.710	ENT WAT BMEP 21.09 31.63	BSFC 0.8286 0.6696	INJ 24.0 22.0	ID 1.72 1.60	AIR TEM V.EFF 68.39 68.28	E.TEMP 491.0 577.4	% OF 1.89 4.29
ENG LD 8 12 16	O.618	FUEL 0.016 0.019 0.024	39.252 32.329 25.831	PERO C.BHF 1.140 1.710 2.280	ENT WAT BMEP 21.09 31.63 42.18	BSFC 0.8286 0.6696 0.6230	INJ 24.0 22.0 22.0	ID 1.72 1.60 1.60	V.EFF 68.39 68.28 67.68	E.TEMP 491.0 577.4 683.6	% OF 1.89 4.28 1.89
ENG LD 8 12 16 20	O.618 0.618 0.617 0.611 0.508	FUEL 0.016 0.019 0.024 0.027	39.252 32.329 25.831 22.416	PERC C.BHF 1.140 1.710 2.280 2.850	ENT WAT BMEP 21.09 31.63 42.18 52.72	BSFC 0.8286 0.6696 0.5715	INJ 24.0 22.0 22.0 22.0	ID 1.72 1.60 1.60 1.56	V.EFF 68.39 68.28 67.68 67.35	E.TEMP 491.0 577.4 683.6 780.9	% OF 1.89 4.29 1.89 1.89
ENG LD 8 12 16	O.618	FUEL 0.016 0.019 0.024	39.252 32.329 25.831	PERO C.BHF 1.140 1.710 2.280	ENT WAT BMEP 21.09 31.63 42.18	BSFC 0.8286 0.6696 0.6230	INJ 24.0 22.0 22.0	ID 1.72 1.60 1.60	V.EFF 68.39 68.28 67.68 67.35	E.TEMP 491.0 577.4 683.6	% OF 1.89 4.28 1.89
ENG LD 8 12 16 20 24	AIR 0.618 0.617 0.611 0.596	FUEL 0.016 0.019 0.024 0.027 0.031	39.252 39.252 32.329 25.831 22.416 19.159	PERC C.BHF 1.140 1.710 2.280 2.850 3.419	BMEP 21.09 31.63 42.18 52.72 63.26	ER = 30 BSFC 0.8286 0.6696 0.6230 0.5715 0.5457	INJ 24.0 22.0 22.0 22.0 23.0	ID 1.72 1.60 1.60 1.56 1.48	V.EFF 68.39 68.28 67.68 67.35 65.93	E.TEMP 491.0 577.4 683.6 780.8 867.2	% OF 1.89 4.78 1.89 1.89 9.25
ENG LD 8 12 16 20 24	AIR 0.618 0.617 0.611 0.596	FUEL 0.016 0.019 0.024 0.027 0.031	39.252 39.252 32.329 25.831 22.416 19.159	PERC C.BHF 1.140 1.710 2.280 2.850 3.419	BMEP 21.09 31.63 42.18 52.72 63.26	ER = 30 BSFC 0.8286 0.6696 0.6230 0.5715 0.5457	INJ 24.0 22.0 22.0 22.0 23.0	ID 1.72 1.60 1.60 1.56 1.48	V.EFF 68.39 68.28 67.68 67.35 65.93	E.TEMP 491.0 577.4 683.6 780.8 867.2	% OF 1.89 4.78 1.89 1.89 9.25
ENG LD 8 12 10 20 24 ENG	AIR 0.618 0.617 0.611 0.508 0.596	FUEL 0.016 0.019 0.024 0.027 0.031	39.252 39.252 32.329 25.831 22.416 19.159	PERC C.BHF 1.140 1.710 2.280 2.850 3.419	BMEP 21.09 31.63 42.18 52.72 63.26 ENT WAT	ER = 30 BSFC 0.8286 0.6696 0.5230 0.5715 0.5457 ER = 30 BSFC	INJ 24.0 22.0 22.0 22.0 23.0	1.72 1.60 1.56 1.56 1.48	V.EFF 68.39 68.28 67.68 67.35 65.93 AIR TEM V.EFF	E.TEMP 491.0 577.4 683.6 780.9 867.2	% OF 1.89 4.28 1.89 1.89 9.25
ENG LD 8 12 10 20 24 ENG LD	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554	FUEL 0.016 0.027 0.021 EED = 10 FUEL 0.016	39.252 39.252 32.329 25.831 22.416 19.159	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHF	BMEP 21.09 31.63 42.18 52.72 63.26 ENT WAT	ER = 30 BSFC 0.8286 0.8286 0.5715 0.5715 0.5457 ER = 30 BSFC 0.8144	INJ 1NJ 24.0 22.0 22.0 22.0 23.0 1 INJ 18.5	NTAKE ID 1.72 1.60 1.60 1.56 1.48 ====== NTAKE ID 1.36	AIR TEM V.EFF 68.39 68.28 67.35 65.93 AIR TEM V.EFF 61.92	E.TEMP 491.0 577.4 683.6 780.9 967.2 P.= 302 E.TEMP 584.6	% OF 1.89 4.28 1.89 9.25 F % OF 1.89
ENG LD 8 12 160 20 24 ENG LD 8	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554 0.550	FUEL 0.016 0.027 0.021 EED = 10 FUEL 0.016 0.019	A/F 39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 28.503	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHF 1.144 1.716	BMEP 21.09 31.60 42.18 52.70 63.26 ENT WAT BMEP 21.16 71.75	ER = 30 BSFC 0.8286 0.6698 0.6230 0.5715 0.5457 ER = 30 BSFC 0.8144 0.6745	INJ 24.0 22.0 22.0 22.0 23.0 INJ INJ 18.5 18.5	10 1.72 1.60 1.60 1.56 1.48 NTALE ID 1.36	V.EFF 68.39 68.28 67.35 65.93 AIR TEM V.EFF 61.92 61.54	E.TEMP 491.0 577.4 683.6 780.8 867.2 F.= 302 E.TEMP 584.6 698.0	% OF 1.89 4.28 1.89 9.25 F % OF 1.89 2.69
ENG LD 8 12 10 20 4 ENG LD 8 12 16	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554 0.550 0.544	FUEL 0.016 0.027 0.031 EED = 10 FUEL 0.016 0.019 0.023	A/F 39.252 25.831 22.416 19.159 000 RPM A/F 35.676 28.533 23.175	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 PERC C.BHF 1.144 1.716 2.288	BMEP 21.09 31.60 42.18 52.72 63.26 ENT WAT BMEP 21.16 51.75 42.30	ER = 30 BSFC 0.8286 0.6698 0.6230 0.5715 0.5457 ER = 30 BSFC 0.8144 0.6745 0.6157	INJ 24.0 22.0 22.0 22.0 23.0 23.0 INJ 18.5 19.0	NTAKE ID 1.72 1.60 1.60 1.56 1.48 NTAKE ID 1.36 1.20	V.EFF 68.39 68.28 67.35 65.93 AIR TEM V.EFF 61.92 61.54 60.86	E.TEMP 491.0 577.4 683.6 780.8 867.2 E.TEMP 584.6 698.0 813.2	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 2.69 7.06
ENG LD 82160 LD 82160	AIR 0.618 0.617 0.611 0.598 0.576 INE SPE AIR 0.550 0.544 0.544	FUEL 0.016 0.027 0.021 EED = 10 FUEL 0.016 0.027 0.021 0.023 0.023 0.023 0.029	39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 28.573 27.280	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHF 1.144 1.716 2.288 2.860	ENT WAT BMEP 21.09 71.67 42.18 52.72 63.26 ENT WAT EMEP 21.16 71.75 52.92	ER = 00 BSFC 0.8286 0.6696 0.5715 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6745 0.6157 0.5916	INJ 24.0 22.0 22.0 23.0 25.0 INJ INJ 18.5 18.5 19.0 21.0	NTAKE ID 1.72 1.60 1.56 1.48 TTAKE ID 1.36 1.20 1.20 1.24	V.EFF 68.39 68.28 67.35 65.93 AIR TEM V.EFF 61.92 61.86 60.83	E.TEMP 491.0 577.4 683.6 780.9 867.2 E.TEMP 584.6 698.0 811.2 914.0	% OF 1.89 4.28 1.89 9.25 F % OF 1.89 2.69 7.06 11.24
ENG LD 8116024 ENG LD 812604	AIR 0.618 0.617 0.611 0.508 0.576 EXECUTE SPE AIR 0.554 0.550 0.544 0.544 0.534	FUEL 0.016 0.019 0.027 0.031 FUEL 0.016 0.019 0.023 0.023 0.023	39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 28.533 27.175 19.280 15.585	PERC C.BHF 1.140 1.710 2.280 3.419 9 PERC C.BHF 1.144 1.716 2.288 2.860 3.433	BMEP 21.09 31.63 42.18 52.72 63.26 ENT WAT BMEP 21.16 31.75 42.35 52.95	ER = 30 BSFC 0.8286 0.6230 0.5715 0.5715 0.5457 ER = 30 BSFC 0.8144 0.6745 0.6157 0.5916	INJ 24.0 22.0 22.0 22.0 23.0 23.0 INJ 18.5 19.0 21.0 21.0	NTAKE ID 1.72 1.60 1.56 1.48 NTAKE ID 1.36 1.28 1.20 1.24 1.20	V.EFF 68.39 68.39 67.35 65.93 AIR TEM V.EFF 61.92 61.54 60.86 59.69	E.TEMP 491.0 577.4 683.6 780.8 867.2 IP.= 302 E.TEMP 584.6 698.0 813.2 914.0 1014.8	% OF 1.89 4.28 1.89 9.25 F % OF 1.89 2.69 7.06 11.24
ENG LD 8116014 == 60 LD 81260124 == 60 LD 812601	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554 0.554 0.544	FUEL 0.016 0.019 0.027 0.031 EED = 10 FUEL 0.016 0.019 0.023 0.023 0.029 0.034	A/F 39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 28.573 23.175 19.280 15.585	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHP 1.144 1.716 2.288 2.860 3.433	BMEP 21.09 31.60 42.18 52.70 63.26 ENT WAT BMEP 21.16 32.90 63.50 ===	ER = 00 BSFC 0.8286 0.6698 0.6230 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6745 0.6157 0.5916 0.5985	INJ 24.0 22.0 22.0 22.0 23.0 23.0 INJ 18.5 19.0 21.0 21.0	NTAKE ID 1.72 1.60 1.60 1.56 1.48 ***********************************	V.EFF 68.39 68.28 67.68 67.35 65.93 AIR TEM V.EFF 61.92 61.54 60.86 59.69	E.TEMP 491.0 577.4 683.6 780.8 867.2 MP.= 302 E.TEMP 584.6 698.0 813.2 914.0 1014.8	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 7.06 11.24 14.07
ENG LD 8116014 == 60 LD 81260124 == 60 LD 812601	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554 0.554 0.544	FUEL 0.016 0.019 0.027 0.031 EED = 10 FUEL 0.016 0.019 0.023 0.023 0.029 0.034	39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 28.533 27.175 19.280 15.585	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHP 1.144 1.716 2.288 2.860 3.433	BMEP 21.09 31.60 42.18 52.70 63.26 ENT WAT BMEP 21.16 32.90 63.50 ===	ER = 00 BSFC 0.8286 0.6698 0.6230 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6745 0.6157 0.5916 0.5985	INJ 24.0 22.0 22.0 22.0 23.0 23.0 INJ 18.5 19.0 21.0 21.0	NTAKE ID 1.72 1.60 1.60 1.56 1.48 ***********************************	V.EFF 68.39 68.28 67.68 67.35 65.93 AIR TEM V.EFF 61.92 61.54 60.86 59.69	E.TEMP 491.0 577.4 683.6 780.8 867.2 MP.= 302 E.TEMP 584.6 698.0 813.2 914.0 1014.8	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 7.06 11.24 14.07
ENG LD 811604 ENG LD 821604 ENG CD 11604 ENG	AIR 0.618 0.617 0.611 0.508 0.576 AIR 0.554 0.5544 0.5544 0.5544 AIR AIR	FUEL 0.016 0.019 0.027 0.031 FUEL 0.016 0.019 0.023 0.023 0.023 0.024	39.252 32.329 25.831 22.416 19.159 300 RFM A/F 35.676 28.577 27.280 15.585	PERC C.BHF 1.140 1.710 2.280 3.419 9 PERC C.BHF 1.144 1.716 2.288 2.860 3.433 10 PERC C.BHP	ENT WAT BMEP 21.09 31.60 42.18 52.72 60.26 ENT WAT BMEP 21.16 42.30 52.50 ENT WAT BMEP	BSFC 0.8286 0.6696 0.6230 0.5715 0.5715 0.5457 ER = 30 BSFC 0.8144 0.6745 0.6157 0.5985 ER = 45 BSFC	INJ 24.0 22.0 22.0 22.0 23.0 INJ 18.5 19.0 21.0 INJ	NTAKE ID 1.72 1.60 1.56 1.48 NTAKE ID 1.36 1.20 1.20 1.20 1.10 INTAKE ID	V.EFF 68.39 67.35 65.93 AIR TEM V.EFF 61.92 60.85 59.69 AIR TEM V.EFF	E.TEMP 491.0 577.4 683.6 780.9 867.2 E.TEMP 584.6 698.0 913.0 1014.8 IF.= 86 F	% OF 1.89 4.28 1.89 9.25 F % OF 1.89 2.69 7.06 11.24 14.00
ENG LD 824004 ENG LD 821604 ENG LD 821604 ENG LD ENG LD	AIR 0.618 0.617 0.611 0.508 0.576 INE SPE AIR 0.554 0.5544 0.5544 0.5544 0.5544 0.5544 0.5564	FUEL 0.016 0.019 0.027 0.031 FUEL 0.016 0.019 0.023 0.023 0.023 0.024 ED = 10 FUEL 0.017	39.252 32.329 25.831 22.416 19.159 300 RFM A/F 35.676 28.573 27.280 15.585 300 RFM A/F	PERC C.BHF 1.140 1.710 2.280 3.419 9 PERC C.BHF 1.144 1.716 2.288 2.860 3.433 10 PERC C.BHP	ENT WAT BMEP 21.09 31.60 42.18 52.72 60.26 ENT WAT BMEP 21.16 42.30 52.50 ENT WAT BMEP 21.17	ER = 30 BSFC 0.8286 0.6696 0.5715 0.5715 0.57457 ER = 30 BSFC 0.8144 0.6745 0.6157 0.5985 ER = 45 BSFC 0.8655	INJ 24.0 22.0 22.0 22.0 23.0 INJ 18.5 19.0 21.0 INJ INJ 29.0	NTAKE ID 1.72 1.60 1.56 1.48 NTAKE ID 1.36 1.20 1.20 1.20 1.20 1.20 1.20 1.20 1.20	AIR TEM V.EFF 68.39 67.35 65.93 AIR TEM V.EFF 61.92 60.85 59.69 AIR TEM V.EFF 71.24	E.TEMP 491.0 577.4 683.6 780.9 867.2 E.TEMP 584.6 698.0 913.0 1014.8 E.TEMP 415.4	% OF 1.89 4.28 1.89 9.25 F % OF 1.89 2.69 7.06 11.24 14.00
ENG LD 824004 ENG LD 821604 ENG LD 821604 ENG LD ENG LD	INE SPE AIR 0.618 0.617 0.611 0.596 INE SPE AIR 0.554 0.544 0.544 0.544 0.544 0.544 0.544 0.544	FUEL 0.016 0.019 0.024 0.027 0.031 EED = 10 FUEL 0.016 0.019 0.023 0.023 0.024 EED = 10 FUEL	A/F 39.252 32.329 25.831 22.416 19.159 300 RPM A/F 35.676 23.175 19.280 15.585 300 RFM A/F 38.549 32.295	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 9 PERC C.BHP 1.144 1.716 2.288 2.860 3.433 10 PERC C.BHP	ENT WAT BMEP 21.09 31.60 42.18 52.70 63.26 ENT WAT BMEP 21.165 42.300 63.50 ENT WAT BMEP 21.17 77	ER = 00 BSFC 0.8286 0.6698 0.6230 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6745 0.6157 0.5916 0.5985 ER = 45 BSFC 0.8655 0.6851	INJ 24.0 22.0 22.0 22.0 23.0 23.0 INJ 18.5 19.0 21.0 21.0 INJ 29.0 26.0	NTAKE ID 1.72 1.60 1.60 1.56 1.48 NTAKE ID 1.36 1.20 1.24 1.20 NTAKE ID 2.96 3.04	AIR TEM V.EFF 68.39 67.35 67.35 65.93 AIR TEM V.EFF 61.92 60.86 60.87 59.69 AIR TEM V.EFF 71.24	E.TEMP 491.0 577.4 683.6 780.8 867.2 E.TEMP 584.6 698.0 813.2 914.0 1014.8 E.TEMP 415.4	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 7.06 11.24 14.07 % OF
ENG LD 821604 # 6 LD 8216	INE SPE AIR 0.618 0.617 0.611 0.596 INE SPE AIR 0.554 0.544 0.544 0.574 INE SPE AIR 0.6571 0.6571	FUEL 0.016 0.019 0.024 0.027 0.031 ED = 10 FUEL 0.016 0.019 0.023 0.023 0.074 ED = 10 FUEL 0.017 0.020 0.023	000 RPM A/F 39.2529 25.851 22.416 19.159 000 RPM A/F 38.576 25.175 19.280 15.585 000 RFM A/F 38.549 52.290	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 PERC C.BHF 1.144 1.716 2.288 2.860 3.433 10 PERC C.BHP	ENT WAT BMEP 21.09 31.63.26 53.26 ENT BMEP 21.1753.50 ENT BMEP 21.1753.73 ENT BMEP 21.1753.73	ER = 00 BSFC 0.8286 0.6698 0.6200 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6157 0.5916 0.5985 ER = 45 BSFC 0.8655 0.6851 0.6022	INJ 24.0 22.0 22.0 22.0 23.0 21.0 INJ 18.5 19.0 21.0 21.0 INJ 29.0 24.0	NTAKE ID 1.72 1.60 1.56 1.48 NTAKE ID 1.36 1.20 1.24 1.20 NTAKE ID 2.96 2.96	AIR TEM V.EFF 68.39 67.35 65.93 AIR TEM V.EFF 61.92 60.83 59.69 AIR TEM V.EFF 71.24 70.25	E.TEMP 491.0 577.4 683.6 780.8 867.2 E.TEMP 584.6 698.0 913.2 914.0 1014.8 E.TEMP 415.4 489.2 579.2	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 7.05 11.24 14.00 % OF
ENG LD 82404 #FG LD 8240 #FN D 114004 #FN D 114004 #FN D 11400	INE SPE AIR 0.618 0.617 0.608 0.596 INE SPE AIR 0.554 0.544 0.544 0.544 0.544 0.657 0.627 0.617	FUEL 0.016 0.019 0.024 0.027 0.031 ED = 10 FUEL 0.016 0.019 0.023 0.023 0.074 ED = 10 FUEL 0.017 0.020 0.023	A/F 39.2529 25.831 22.416 19.159 25.676 25.677 27.280 15.585 20.0 RFM A/F 38.549 32.290 27.410	PERC C.BHF 1.140 1.710 2.280 2.850 3.419 PERC C.BHF 1.144 1.716 2.860 3.433 10 PERC C.BHP	ENT WAT BMEP 21.09 31.05 42.12 42.72 63.26 ENT WAT BMEP 21.175 23.90 ENT BMEP 21.776 21.776 22.94	ER = 00 BSFC 0.8286 0.6698 0.6200 0.5715 0.5457 ER = 00 BSFC 0.8144 0.6157 0.5916 0.5985 ER = 45 BSFC 0.8655 0.6851 0.6022	INJ 24.0 22.0 22.0 22.0 23.0 24.0 21.0 21.0 21.0 21.0 21.0 21.0 21.0 21	NTAKE ID 1.70 1.60 1.56 1.48 NTAKE ID 1.36 1.20 1.24 1.20 NTAKE ID 2.96 2.64	AIR TEM V.EFF 68.39 67.35 67.35 65.93 AIR TEM V.EFF 61.92 60.86 60.87 59.69 AIR TEM V.EFF 71.24	E.TEMP 491.0 577.4 683.6 780.8 867.2 E.TEMP 584.6 698.0 817.2 914.0 1014.8 E.TEMP 415.4 489.2 556.6	% OF 1.89 4.28 1.89 1.89 9.25 F % OF 1.89 2.69 7.06 11.24 14.07

			OQ RPM								F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E. TEMP	% OP
8 12 16 20 24	0.607 0.605			1.142 1.713 2.284 2.855 3.426	31.69 42.25 52.82	0.6258 0.5768	23.5	2.56 2.32 2.00	68.00 67.50 67.30 65.64 65.64		1.89 1.89 1.89 3.28 5.67
			00 RPM								F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
12 16 20 24	0.561 0.552 0.553 0.547	0.020 0.024 0.028 0.034	33.518 28.240 23.016 19.595 15.886	1.712 2.283 2.854 3.425	31.68 42.24 52.80 63.36	0. 5 931 0.6030	21.5 21.5 21.0 21.5	1.76 1.60 1.56 1.56	62.35 62.34 61.38 61.47 60.80	746.6 847.4 944.6	1.89 1.89 1.89 1.89 8.85
			00 RPM								
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
8 12 16 20 24	1.471 1.471 1.426 1.414 1.410	0.034 0.040 0.047 0.054 0.061	41.577 35.514 30.538 26.225 22.966	3.413 4.551 5.688	31.57 42.10 52.62	0.7084 0.6158 0.5687	31.0 31.0 32.0	0.92 0.96 0.92	78.78 78.78 78.52 77.81 77.57	662.0 741.2 833.0	10.25 14.23
ENG	INE SPE	ED = 20	*====== 00 RPM	14 PERC	== ENT WAT	#==#=== ER = 0	===== IN	TAKE A	IR TEMP	: .= 167 F	.
	INE SFE		00 RPM A/F			#==#== ER = 0 BSFC				: .= 167 F E.TEMP	
בם	AIR	FUEL 0.035	A/F 37.808 32.488 27.500 23.595	C.BHF 2.276 3.414	BMEP 21.06	BSFC 0.9256 0.7122 0.6278 0.5805	INJ 31.0	ID 0.92 0.88 0.88 0.84	V.EFF 73.16 72.55 72.18 71.58	E.TEMP 669.2 743.0	% DP 5.87 6.85 7.86 11.84
2D 8 12 16 20 24	AIR 1.728 1.717 1.710 1.299 1.708	FUEL 0.035 0.041 0.048 0.055 0.064	A/F 37.808 32.488 27.500 23.595 20.464	C.BHF 2.276 3.414 4.552 5.691 6.829	BMEP 21.06 31.58 42.11 52.64 63.17	BSFC 0.9256 0.7122 0.6278 0.5805 0.5616	INJ 31.0 31.0 31.0 31.0 32.5	ID 0.92 0.88 0.88 0.84 0.76	V.EFF 73.16 72.55 72.18 71.58 72.08	669.2 743.0 829.4 933.8 1031.0	% DF 5.87 6.86 7.86 11.84 37.71
8 12 16 20 24 ### ENG	AIR 1.728 1.717 1.710 1.299 1.708	FUEL 0.035 0.041 0.048 0.055 0.064	A/F 37.808 32.488 27.500 23.595 20.464	C.BHF 2.276 3.414 4.552 5.691 6.829 15 FERC	BMEP 21.06 31.58 42.11 52.64 63.17	BSFC 0.9256 0.7122 0.6278 0.5805 0.5616	INJ 31.0 31.0 31.0 31.0 32.5	ID 0.92 0.88 0.88 0.84 0.76	V.EFF 73.16 72.55 72.18 71.58 72.08	669.2 743.0 829.4 933.8 1031.0	% DF 5.87 6.86 7.86 11.84 37.71
D 811604 = RG D 82160	AIR 1.728 1.717 1.710 1.299 1.708 INE SPE AIR 1.186 1.186 1.184 1.176	FUEL 0.035 0.041 0.055 0.064 ED = 20 FUEL 0.034 0.049 0.049 0.057	A/F 37.808 32.488 27.500 23.595 20.464 	C.BHP 2.276 3.414 4.552 5.691 6.829 15 FERC C.BHP 2.278 3.417 4.555 5.694	BMEP 21.06 31.58 42.11 52.64 63.17 ENT WAT BMEP 21.07 31.01 42.14 52.68	BSFC 0.9256 0.7122 0.6278 0.5805 0.5616 ER = 0 BSFC 0.8863 0.7150 0.6410 0.6040	INJ 31.0 31.0 31.0 31.0 32.5 IN INJ 32.5 32.5 32.5	1D 0.92 0.88 0.84 0.76 TAKE A 1D 0.92 0.88 0.76	V.EFF 73.16 72.55 72.18 71.58 72.08 72.08 V.EFF 65.47 65.37 64.93	E.TEMP 669.2 743.0 829.4 933.8 1031.0 7.= 302 F E.TEMP 723.2 829.4 960.8 1068.8	% DF 5.87 6.86 7.86 11.84 37.71 % OF 6.86 8.66 12.64 16.82
ENG L 874004	AIR 1.728 1.717 1.710 1.279 1.708 INE SPE AIR 1.186 1.190 1.184 1.176	FUEL 0.035 0.041 0.055 0.064 ED = 20 FUEL 0.034 0.041 0.049 0.057 0.072	A/F 37.808 32.488 27.500 23.595 20.464 30.86M A/F 35.249 28.249 29.519	C.BHF 2.276 3.414 4.552 5.691 6.829 15 FERC C.BHP 2.278 3.41/ 4.555 5.694 6.804	BMEP 21.06 51.58 42.11 52.64 63.17 ENT WAT BMEP 21.07 51.61 42.14 52.68 63.21	BSFC 0.9256 0.7122 0.6278 0.5805 0.5616 ER = 0 BSFC 0.8863 0.7150 0.6410 0.6281	INJ 31.0 31.0 31.0 32.5 IN INJ 32.5 32.5 32.5 32.5	1D 0.92 0.88 0.84 0.76 TAKE A 1D 0.92 0.88 0.76 0.76	V.EFF 73.16 72.18 71.58 72.08 72.08 V.EFF 45.47 45.15 45.39 44.93	E.TEMP 669.2 743.0 829.4 933.8 1031.0 7.= 302 F E.TEMP 723.2 829.4 960.8 1068.8 1247.0	% OP 5.87 6.86 7.86 11.84 37.71 % OF 6.86 8.06 12.64 16.82 49.05
ENG L 874004	AIR 1.728 1.717 1.710 1.299 1.708 INE SPE AIR 1.186 1.186 1.190 1.184 1.176	FUEL 0.035 0.041 0.055 0.064 EED = 20 FUEL 0.034 0.041 0.049 0.057 0.072	A/F 37.808 32.488 27.500 23.595 20.464 ==================================	C.BHF 2.276 3.414 4.552 5.691 6.829 15 FERC C.BHP 2.278 2.417 4.555 5.694 6.804	BMEP 21.06 31.58 42.11 52.64 63.17 ENT WAT BMEP 21.07 31.01 42.14 52.68 63.21 ENT WAT	BSFC 0.9256 0.7122 0.6278 0.5805 0.5616 ===================================	INJ 31.0 31.0 31.0 31.0 32.5 IN INJ 32.5 32.5 32.5 32.5	1D 0.92 0.88 0.84 0.76 TAKE A 1D 0.92 0.88 0.76 0.68	V.EFF 73.16 72.55 72.18 71.58 72.08 72.08 V.EFF 65.47 65.15 65.39 64.93 64.92	E.TEMP 669.2 743.0 829.4 933.8 1031.0 7.= 302 F E.TEMP 723.2 829.4 960.8 1068.8 1247.0	% DP 5.87 6.86 7.86 11.84 37.71 % OF 6.86 8.06 12.64 16.82 49.05

ENG			00 RPM								F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	v.EFF	E.TEMP	% OF
8 12 16 20 24	1.331 1.336 1.316 1.304 1.298		33.038	3.429 4.572	21.15 31.72 42.29 52.87 63.43	0.7074 0.6241 0.5859	28.0 28.0 28.0		73.4 5 72.73	714.2 811.4	8.46 10.85
			OO RPM								F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
12 16 20 24	1.160 1.150 1.141 1.129	0.041 0.049 0.059 0.073	23.381 19.495 15.433	3.436 4.581 5.727 6.877	31.78 42.38 52.97 63.61	0.6382	28.5 28.5 28.5 28.5	0.84 0.80 0.80 0.80	64.52 64.05 63.47	833.0 957.2 1108.4 1259.6	1.89 9.65 27.56
=== ENG	INE SPE	ED = 20	00 RPM	19 PERC	== ENT WAT	======= ER = 30	≈===== I	TTAKE	AIR TEM	P.= 88 F	=
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	v.EFF	E.TEMP	% OP
8 12 16 20 24	1.398 1.424 1.410 1.404 1.387	0.033 0.038 0.045 0.052 0.059	42.601 37.346 31.086 27.093 23.471	3.442 4.590 5.737		0.6648 0.5930 0.5418	30.0	2.04 1.96	78.78 80.31 79.54 79.18 78.24	514.4 590.0 680.0 761.0 861.8	1.89 1.89 4.28 4.28 5.07
===											
ENG			00 RPM			======= ER = 30					F
ENG LD			00 RPM			ER = 30	I		AIR TEM		
LD	INE SPE AIR 1.309	ED = 20	00 RPM A/F 39.149 33.110	PERC C.BHP 2.292 3.438 4.584 5.730	ENT WAT BMEP 21.20 31.80 42.40 53.00	ER = 30	I INJ 29.0 29.0	NTAKE	AIR TEM	F.= 167 E.TEMP 591.8 680.0 760.8 888.8	% OF 1.89 1.89 1.89
ED 8 12 16 20 24	AIR 1.309 1.303 1.301 1.303 1.278	FUEL 0.033 0.039 0.046 0.054 0.062	A/F T9.149 T3.110 28.044 24.036 20.540	PERC C.BHP 2.292 3.438 4.584 5.730 6.876	ENT WAT BMEP 21.20 31.80 42.40 53.00 63.60	ER = 30 RSFC 0.8756 0.6869 0.6073 0.5677 0.5430	INJ 29.0 29.0 27.0 27.0 26.5	ID 2.04 1.92 1.92 1.56 1.52	V.EFF 73.59 73.25 73.12 73.25 71.84	E.TEMP 591.8 680.0 780.8 888.8 996.8	% OF 1.89 1.89 1.89 5.87 12.87
ED 8 12 16 20 24 ENG	AIR 1.309 1.303 1.301 1.303 1.278	FUEL 0.033 0.039 0.046 0.054 0.062	A/F 39.149 33.110 28.044 24.036 20.540	PERC C.BHP 2.292 3.438 4.584 5.730 6.876	ENT WAT BMEP 21.20 31.80 42.40 53.00 63.60 ENT WAT	ER = 30 RSFC 0.8756 0.6869 0.6073 0.5677 0.5430	INJ 29.0 29.0 27.0 27.0 26.5	ID 2.04 1.92 1.92 1.56 1.52	V.EFF 73.59 73.25 73.12 73.25 71.84	E.TEMP 591.8 680.0 780.8 888.8 996.8	% OF 1.89 1.89 5.87 12.87
LD 8216024 == G LD 82160	INE SPE AIR 1.309 1.303 1.301 1.303 1.278 ELECTRICAL SPE AIR 1.171 1.157 1.147	FUEL 0.033 0.079 0.046 0.054 0.062 FUEL 0.034 0.054 0.054 0.050 0.050 0.050	A/F 39.149 33.110 28.044 24.036 20.540	PERC C.BHP 2.4384 5.730 6.876 21 PERC C.BHF 2.4374 5.724	ENT WAT BMEP 21.20 31.80 42.40 53.00 63.60 ENT WAT BMEP 21.18 31.77 42.394	ER = 30 BSFC 0.8756 0.6869 0.6073 0.5677 0.5430 ER = 30 BSFC 0.8990 0.7743 0.6548 0.6274	INJ 29.0 27.0 27.0 26.5 INJ 27.0 26.0 26.0 26.0	NTAKE ID 2.04 1.92 1.92 1.56 1.52 NTAKE ID 1.00 1.00 0.92	AIR TEM V.EFF 73.59 73.12 73.25 71.84 AIR TEM V.EFF 65.64 64.47 64.02	E.TEMP 591.8 680.0 780.8 888.8 996.8 F.= TOZ E.TEMP 704.0 836.6 957.2 1106.6	% OP 1.89 1.89 1.89 5.81 12.81 F % OP 1.89 2.89 4.48 19.80
ED 8 126024 ENG LD 8 12504 ENG CD 24 ENG CD 8 12504	INE SPE AIR 1.309 1.303 1.301 1.303 1.278 INE SPE AIR 1.171 1.159 1.151 1.147 1.178	FUEL 0.033 0.079 0.046 0.054 0.062 EED = 20 FUEL 0.034 0.050 0.040 0.074	79.149 79.149 73.110 28.044 24.076 20.540 70.000 RFM A,F 74.148 27.580 27.026 19.092	PERC C. BHP 2.4384 4.5844 5.730 6.81 PERC C. BHF 2.4579 5.868	ENT WAT BMEP 21.20 31.80 42.40 53.00 63.60 == ENT WAT BMEP 21.18 77 42.76 52.94 63.50	ER = 30 BSFC 0.8756 0.6875 0.5677 0.5430 ER = 30 BSFC 0.8990 0.7343 0.6548 0.6274 0.6488	INJ 29.0 29.0 27.0 27.0 26.5 INJ 27.0 26.0 26.0 26.0	NTAKE ID 2.04 1.92 1.92 1.56 1.52 NTAKE ID 1.00 1.00 0.92 0.92	AIR TEM V.EFF 73.25 73.25 73.25 71.84 AIR TEM V.EFF 65.64 64.95 64.47 64.02 63.79	E.TEMP 591.8 680.0 780.8 888.8 996.8 IP.= T02 E.TEMP 774.0 836.6 957.2 1106.6 1230.8	% OP 1.89 1.89 1.89 5.87 12.87 % OP 1.89 2.89 4.48 19.80 41.09
LD 821604 = RG LD 821604 = RG	INE SPE AIR 1.309 1.301 1.301 1.303 1.278 INE SPE AIR 1.171 1.159 1.151 1.147 1.178	FUEL 0.033 0.039 0.046 0.054 0.062 ED = 20 FUEL 0.034 0.050 0.040 0.074	79.149 73.110 28.044 24.036 20.540 7.540 7.580 27.580 27.026 19.092 15.728	PERC C.BHP 2.438 4.584 5.730 6.876 21 PERC C.BHF 2.4374 4.5724 6.88 PERC	ENT WAT BMEP 21.20 31.80 42.40 55.00 63.60 ENT WAT BMEP 21.18 31.77 42.76 52.94 63.50 ENT WAT	ER = 30 BSFC 0.8756 0.6869 0.6077 0.5677 0.5430 ER = 30 BSFC 0.8990 0.7743 0.6548 0.6274 0.6488 ==================================	INJ 29.0 29.0 27.0 27.0 26.5 INJ 27.0 26.0 26.0 26.0	NTAKE 1D 2.04 1.92 1.92 1.56 1.52 NTAKE 1D 1.00 1.00 0.92 0.92 NTAKE	AIR TEM V.EFF 73.25 73.25 73.25 71.84 AIR TEM V.EFF 65.64 64.47 64.02 63.79 AIR TEM	E.TEMP 591.8 680.0 780.8 888.8 996.8 IP.= T02 E.TEMP 774.0 836.6 957.2 1106.6 1230.8	% OP 1.89 1.89 1.89 5.87 12.87 F % CP 1.89 2.89 4.48 19.80 41.09

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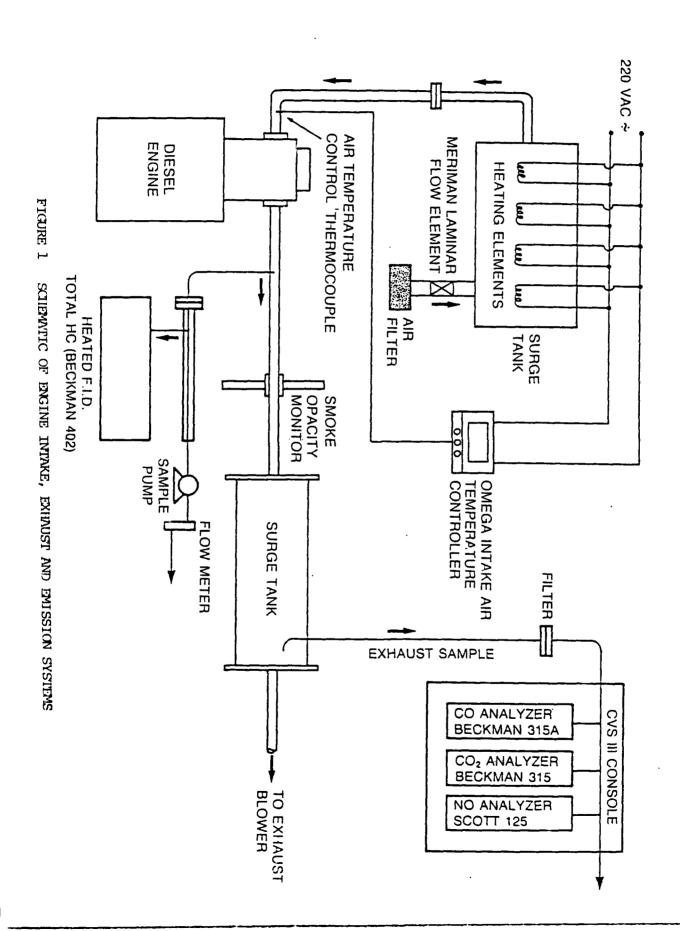
				- -					_		
ENGINE SPEED = 2000 RPM			PERC	ENT WAT	ER = 45	I	NTAKE	AIR TEM	F.= 167	F	
ĻΦ	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
8	1.347	0.036	37.447	2.284	21.13	0.9448	33.0	2.00	74.92	579.2	1.89
12	1.345	0.041	32.634	3.426	31.69	0.7216	33.0	2.00	74.80	645.8	1.89
16	1.338	0.046	29.025	4.568	42.25	0.6056	32.0	1.92	74.44	714.2	1.89
20	1.336	0.055	24.085	5.710	52.82	0.5828	29.0	1.76	74.31	814.8	3.88
24	1.331	0.063	21.075	6.851	63.38	0.5533	29.0	1.64	74.07	930.2	5.87
#======================================				24	==	=======	*****	*====	******		
ENG	INE SPE	ED = 20	OO RPM	PERC	ENT WAT	ER = 45	I	NTAKE	AIR TEM	P.= 302	F
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	D	V.EFF	E.TEMP	% OF
8	1.232	0.005	34.797	2.277	21.07	0.9324	25.5	1.96	67.97	707.0	5.87
12	1.219	0.043	28.629	3.416	31.60	0.7476	25.5	1.84	67.26	807.8	8.85
16	1.208	0.049	24.495	4.555	42.14	0.6496	25.0	i.84	66.67	917.6	9.85
20	1.206	0.040	20.251	5.694	52.67	0.6278	25.0	1.72	66.58	1045.4	9.85
24	1.189	0.074	16.164	6.832	63,20	0.6462	25.0	1.68	65.64	1189.4	22.79
2==	*****		======================================	25	==	3555555	=====	=====	======		
ENG	INE SPE	ED = 10	00 RPM	PERC	ENT WAT	ER = 0	IN	TAKE A	IR TEMP	.≃ 88 F	
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OP
16	0.680	0.022	30.728	2.264	41.88	0.5864	23.0	0.00	73.74	324.0	0.00
15	0.480	0.022	30.703	2.264	41.88	0.5869	23.0	0.00	73.74	324.0	0.00
16	0.680	0.022	30.796	2.264	41.88	0.5851	23.0	0.00	73.74	324.0	0.00
15	0.680	0.022	30.678	2.264	41.88	0.5873	23.0	0.00	73.74	324.0	0.00
20	0.673	0.026	25.965	2.831	52,37	0.5521	23.0	0.00	73.42	377.0	0.00
20	0.675	0.026	25.977	2.831	52,37	0.5518	23.0	0.00	73.42	377.0	0.00
20	0.676	0.026	25.977	2.831	52,37	0.5518	23.0	0.00	73.42	377.0	0.00
20	0.676	0.026	26.039	2.851	52,37	0.5505	23.0	0.00	73.42	377.0	0.00
24	0.674	0.001	21.900	3.397	62,85	0.5435	25.0	0.00	73.17	401.0	0.00
24	0.674	0.031	21.976	3.397	62,85	0.5417	23.0	0.00	73.17	431.0	0.00
24	0.674	0.031	22.043	3.397	62.85	0.5401	25.0	0.00	73.17	431.0	0.00
24	0.674	0.031	21.953	3.397	62,85	0.5428	23.0	0.00	73.17	401.0	0.00
				_							
ENGINE SPEED = 1000 RPM PER											
E140	THE SEE	.ED = 10	OO RPM	PERL	ENI WAI	EK = 15	1	NIAKE	AIR IEM	F.= 88 F	•
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E.TEMP	% OF
16	0.672	0.022	29.885	2.267	41.94	0.5951	24.0	0.00	73.13	317.0	0.00
16	0.672	0.023	29.833	2.267	41,94	0.5961	24.0	0.00	73.15	517.0	
15	0.672	0.023	29.818	2.267	41.94	0.5964	24.0	0.00	73.13	317.0	0.00
16	0.672	0.023	29.818	2.267	41.94	0.5964	24.0	0.00	73.13	317.0	0.00
20	0.672	0.027	25.046	2.800	52.42	0.5681	24.0	0.00	73.13	371.0	
20	0.672	0.027	25.046	2.855	52.42	0.5681	24.0	0.00	73.13	371.0	
20	0.672	0.027	25.061	2.853	52.42	0.5677	24.0	0.00	73.13	371.0	0.00
20	0.672	0.027	25.179	2.800	52.42	0.5651	24.0	0.00	73.13	371.0	0.00
24	0.672	0.071	21,494	3.400	62.90	0.5516	24.0	0.00	70.10	425.0	0.00
24	0.672	0.031	21.457	3.400	62.90	0.5526	24.0	0.00	73.13	425.0	0.00
24	0.672	0.031	21.479	7.400	62.90	0.5520	24.0	0.00	73.13	426.0	0.00
24	0.672	0.001	21.465	3.400	62,90	0.5524	24.0	0.00	73.13	426.0	0.00
			-1.760	⇒ • • • • • • • • • • • • • • • • • • •	O2. ///		U	10 a 10 10 1	7 and 4 and	- ZO - W	40 a 50 60

ENGINE SPEED = 1000 RPM					27 *====================================							
ENG	INE SPE	ED = 10	OO RPM	PERC	EN! WA!	EK = 30	1	MIMME	HIK IEN	50 1		
בט	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E. TEMP	% OP	
	s . 70		20.425	2 2/7	4. 04	0 4104	24.0	0.00	73.13	312.0	0.00	
16	0.672	0.023	29.125	2.267	41.94 41.94	0.6106	24.0	0.00	73.13	312.0	0.00	
16	0.672	0.023	29.144	2.267	41.94	0.6102	24.0	0.00	73.13	312.0	0.00	
16	0.672	0.023	29.144	2.267					73.13	312.0	0.00	
15	0.672	0.023	27.061	2.267	41.94	0.6120	24.0	0.00			0.00	
20	0.672	0.027	24.747	2.833	52.42	0.5749	24.0	0.00	73.13	371.0		
20	0.672	0.027	24.710	2.833	52.42	0.5758	24.0	0.00	73.13	371.0	0.00	
20	0.672	0.027	24.720	2.833	52.42	0.5756	24.0	0.00	73.13	371.0	0.00	
20	0.672	0.027	24.701	2.833	52.42	0.5760	24.0	0.00	73.13	371.0	0.00	
24	0.672	0.032	21.273	3.400	52.90	0.5574	24.0	0.00	73.13	428.0	0.00	
24	0.672	0.032	21.291	3.400	62.90	0.5569	24.0	0.00	73.13	428.0	0.00	
24	0.672	0.032	21.263	3.400	62.90	0.5576	24.0	0.00	75.15	428.0	0.00	
24	0.672	0.032	21.162	3.400	62.90	0. 5 603	24.0	0.00	73.13	428.0	0.00	
						TED - A						
,	ENGINE SPEED = 2000 RPM			FERU	PERCENT WATER = 0			INTAKE AIR TEMP.= 88 F				
LD	AIR	FUEL	A/F	C.BHP	BMEP	BSFC	INJ	ID	V.EFF	E. TEMP	% OP	
16	1.491	0.046	32.108	4.546	42.05	0.6130	30.0	0.00	81.83	375.0	0.00	
16	1,491	0.046	32.142	4.546	42.05	0.6123	30.0	0.00	81.83	375.0	0.00	
16	1.491	0.046	32.155	4.546	42.05	0.6121	30.0			375.0		
16	1.491	0.046	32.155	4.546	42.05			0.00	81.83		0.00	
20						0.6121	30.0	0.00	81.83	375.0	0.00	
	1.478	0.054	27.304	5.685	52.59	0.5714	30.0	0.00	81.23	428.0	0.00	
20	1.478	0.054	27.311	5.685	52.59	0.5713	30.0	0.00	81.23	428.0	0.00	
20	1.479	0.054	27.358	5.685	52.59	0.5703	30.0	0.00	81.23	428.0	0.00	
20	1.478	0.054	27.351	5.685	52.59	0.5704	30.0	0.00	81.23	428.0	0.00	
24	1.469	0.062	23.654	6.822	6J.10	0.5464	30.0	0.00	80.74	486.0	0.00	
24	1.469	0.062	23.674	6.822	6J.10	0.5459	30.0	0.00	80.74	486.0	0.00	
24	1.469	0.062	23.674	6.822	63.10	0.5459	30.0	0.00	80.74	486.0	0.00	
24	1.469	0.062	23.714	6.822	63.10	0.5450	30.0	0.00	80.74	486.9	0.00	
			23.714			0.5450	30.0				0.00	
===	******			: 29	==	.=======	30.0		122222			
===	******			: 29	==	.=======	30.0		AIR TEN	: 1P.= 88 F		
ENG LD	GINE SPE	ED = 20 FUEL	000 RPM A/F	: 29 PERC C.BHP	ENT WAT	ER = 15 BSFC	30.0 1 INJ	ID	AIR TEM	: 1P.= 88 F E.TEMP	% OF	
ENG LD	GINE SPS AIR 1.484	FUEL 0.046	000 RPM A/F 32.599	29 PERC C.BHP 4.549	ENT WAT BMEP 42.08	ER = 15 BSFC 0.6004	30.0 I INJ 31.0	ID 0.00	AIR TEN	: MP.= 88 F E.TEMP 356.0	% QP	
ENG LD 16	AIR 1.484	FUEL 0.046	000 RPM A/F 32.599 32.542	29 PERC C.BHP 4.549 4.549	BMEP 42.08 42.08	ESFC 0.6004 0.6015	30.0 INJ 31.0 31.0	ID 0.00 0.00	AIR TEM V.EFF 81.64 81.64	: MP.= 88 F E.TEMP 356.0 356.0	% OP	
ENG LD 16 16	AIR 1.484 1.484 1.484	FUEL 0.046 0.045 0.045	32.599 32.542 32.665	29 PERC C.BHP 4.549 4.549 4.549	BMEP 42.08 42.08 42.08	ER = 15 BSFC 0.6004 0.6015 0.5992	30.0 INJ 31.0 31.0 31.0	ID 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64	: MP.= 88 F E.TEMP 356.0 356.0 356.0	% OF	
ENG LD 16 16	AIR 1.484 1.484 1.484 1.484	FUEL 0.046 0.045 0.046 0.046	000 RPM A/F 32.599 32.542 32.665 32.379	29 PERC C.BHP 4.549 4.549 4.549 4.549	EMEP 42.08 42.08 42.08 42.08 42.08	BSFC 0.6004 0.6015 0.5992 0.6045	30.0 INJ 31.0 31.0 31.0 31.0	ID 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64	: IP.= 88 F E.TEMP 356.0 356.0 356.0 356.0	% OP	
ENG LD 16 16 15 15	AIR 1.484 1.484 1.484 1.484 1.487	FUEL 0.046 0.045 0.046 0.045	32.599 32.542 32.645 32.379 27.928	29 PERC C.BHP 4.549 4.549 4.549 4.549 5.687	ENT WAT BMEP 42.08 42.08 42.08 42.08 52.60	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573	30.0 INJ 31.0 31.0 31.0 31.0 31.0	O.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64	: MP.= 88 F E.TEMP 356.0 356.0 356.0	% OP	
ENG LD 16 16 15 15 20 20	AIR 1.484 1.484 1.484 1.485 1.475	FUEL 0.046 0.045 0.046 0.045 0.050	32.599 32.599 32.542 32.665 32.379 27.928 27.904	29 PERC C.BHP 4.549 4.549 4.549 4.549 5.687	ENT WAT BMEP 42.08 42.08 42.08 42.08 52.60	FER = 15 BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5578	30.0 INJ 31.0 31.0 31.0 31.0 31.5	ID 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64	: IP.= 88 F E.TEMP 356.0 356.0 356.0 356.0	% OP	
ENG LD 16 16 15 15 15 20 20	AIR 1.484 1.484 1.484 1.484 1.475 1.475	FUEL 0.046 0.045 0.045 0.045 0.050 0.050	32.599 32.542 32.665 32.379 27.928 27.904 27.863	29 PERC C.BHP 4.549 4.549 4.549 4.549 5.687	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5578 0.5578	30.0 INJ 31.0 31.0 31.0 51.0 51.5	O.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64 81.64	: 1P.= 88 F E.TEMP 356.0 356.0 356.0 405.0	% OP 0.00 0.00 0.00 0.00	
ENG LD 16 16 16 16 16 16 16 16 16 16 16 16 16	AIR 1.484 1.484 1.484 1.475 1.475 1.475	FUEL 0.046 0.045 0.046 0.045 0.050	32.599 32.542 32.542 32.665 32.379 27.928 27.904 27.863 27.879	29 PERC C.BHP 4.549 4.549 4.549 4.549 5.687	ENT WAT BMEP 42.08 42.08 42.08 42.08 52.60	FER = 15 BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5578	30.0 INJ 31.0 31.0 31.0 31.0 31.5	O.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64 81.15	E.TEMP 356.0 356.0 356.0 356.0 405.0	% OP 0.00 0.00 0.00 0.00 0.00	
ENG LD 16 16 16 16 16 16 16 16 16 16 16 20 20 20 20 20	AIR 1.484 1.484 1.484 1.484 1.475 1.475	FUEL 0.046 0.045 0.045 0.045 0.050 0.050	A/F 32.599 32.542 32.665 32.379 27.928 27.904 27.863 27.879 24.482	29 PERC C.BHP 4.549 4.549 4.549 4.549 5.687	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5578 0.5578	30.0 INJ 31.0 31.0 31.0 51.0 51.5	O.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64 81.15 81.15	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0	% OP 0.00 0.00 0.00 0.00 0.00 0.00	
ENG LD 16 16 16 16 16 16 16 16 16 16 16 16 16	AIR 1.484 1.484 1.484 1.475 1.475 1.475	FUEL 0.046 0.045 0.045 0.050 0.050 0.050	A/F 32.599 32.542 32.665 32.379 27.928 27.904 27.863 27.879 24.482	29 PERC C.BHP 4.549 4.549 4.549 5.687 5.687 5.687	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60 52.60	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5578 0.5586 0.5583	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5	ID 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.64 81.15 81.15	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00	
ENG LD 16 16 16 16 16 16 16 16 16 16 16 16 16	AIR 1.484 1.484 1.484 1.475 1.475 1.475 1.475	FUEL 0.046 0.045 0.045 0.046 0.053 0.053 0.053	A/F 32.599 32.542 32.665 32.379 27.908 27.904 27.863 27.879 24.482 24.522	29 PERC C.BHP 4.549 4.549 4.549 5.687 5.687 6.824	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60 52.60 63.13	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5586 0.5586 0.5583 0.5266 0.5258	30.0 INJ 31.0 31.0 31.0 51.5 51.5 51.5 52.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 81.15 81.15	E.TEMP 354.0 354.0 354.0 354.0 405.0 405.0 405.0 405.0 405.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00	
ENG LD 16 16 15 15 15 20 20 20 20 24 24	AIR 1.484 1.484 1.484 1.475 1.475 1.475 1.475	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.050	A/F 32.599 32.542 32.665 32.379 27.908 27.904 27.863 27.879 24.482 24.522	29 PERC C.BHP 4.549 4.549 4.549 5.687 5.687 5.687 5.687	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60 52.60 52.60 52.60	BSFC 0.6004 0.6015 0.5992 0.6045 0.5573 0.5586 0.5588 0.5588	30.0 INJ 31.0 31.0 51.0 51.5 51.5 51.5 51.5 52.0 52.0	O.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 81.15	E.TEMP 354.0 354.0 354.0 354.0 405.0 405.0 405.0 405.0 405.0	% OP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG 16 16 16 16 16 16 16 16 16 16 16 16 16	AIR 1.484 1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466	FUEL 0.046 0.045 0.045 0.055 0.055 0.055 0.055 0.055 0.060 0.060	32.599 32.542 32.542 32.665 32.379 27.924 27.863 27.879 24.482 24.522 24.522	PERC C.BHP 4.549 4.549 4.549 7.5687 6.8224 6.8224 6.836	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60 52.60 52.60 53.13 63.13 63.13	FER = 15 BSFC 0.4004 0.4015 0.5992 0.4045 0.5578 0.5578 0.5583 0.5268 0.5258	30.0 INJ 31.0 31.0 31.0 31.5 31.5 31.5 32.0 82.0 52.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 81.15 81.15 80.66 80.66	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0	% OP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG 16 16 15 10 20 20 20 20 24 24 24	AIR 1.484 1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466	FUEL 0.046 0.045 0.045 0.055 0.055 0.055 0.055 0.055 0.060 0.060	32.599 32.542 32.542 32.665 32.379 27.904 27.863 27.879 24.482 24.522 24.474 24.522	PERC C.BHP 4.549 4.549 4.549 4.56887 4.66887 6.8624 6.864 6.86	ENT WAT BMEP 42.08 42.08 42.08 52.60 52.60 52.60 53.13 63.13 63.13	FER = 15 BSFC 0.6004 0.6015 0.5992 0.6045 0.5578 0.5586 0.5588 0.5268 0.5258	30.0 INJ 31.0 31.0 31.0 31.5 51.5 51.5 52.0 52.0 52.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.65 81.15 81.15 81.15 80.66 80.66 80.66	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG 16 16 16 16 16 16 16 16 16 16 16 16 16	AIR 1.484 1.484 1.484 1.475 1.475 1.475 1.475 1.475	FUEL 0.046 0.045 0.045 0.053 0.053 0.053 0.060 0.060	32.599 32.542 32.542 32.665 32.379 27.928 27.904 27.863 27.863 27.879 24.482 24.474 24.522	29 PERC C.BHP 4.549 4.549 4.549 5.687 5.687 6.824 6.824 6.824	ENT WATER TO THE TENT WATER TO	FER = 15 BSFC 0.6004 0.6015 0.5992 0.5578 0.5578 0.5583 0.5264 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.5 31.5 31.5 31.5 32.0 32.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 81.15 80.66 80.66 80.66	E.TEMP 354.0 354.0 355.0 355.0 405.0 405.0 462.0 462.0 462.0	% OP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 1645 150 200 204 244 24 24 24 25 ENG LD	1.484 1.484 1.475 1.475 1.475 1.475 1.4666 1.4666 1.4666	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.050 0.060 0.060	A/F 32.599 32.542 32.665 32.379 27.928 27.904 27.863 27.879 24.522 24.522 24.522 24.522	29 PERC C. BHP 4.5499 4.5497 7.74 6.882.24 4.5688.26	ENT WAT BMEP 42.08 422.080 522.6000 522.600 522.600 522.600 522.600 522.600 522.600 522.600 522.600 522.600 52	FER = 15 BSFC 0.4004 0.4015 0.5992 0.4045 0.5578 0.5578 0.5584 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 32.0 32.0 32.0	O.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 81.15 80.66 80.66 80.66	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 1641500000000000000000000000000000000000	AIR 1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466 1.466	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.060 0.060 0.060 FUEL 0.046	32.599 32.542 32.542 32.665 32.379 27.924 27.863 27.879 24.474 24.522 24.522 24.522 24.522	PERC C. BHP 4.549 4.549 4.549 7.56887 6.88224 6.88224 6.88224 6.882 6.88	ENT WAT BMEP 42.08 422.060 522.660 522.660 523.117 EAT BMEP 42.08	FER = 15 BSFC 0.4004 0.4015 0.5992 0.4045 0.5578 0.5578 0.5583 0.5246 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.5 31.5 31.5 32.0 52.0 52.0 INJ	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 81.15 80.66 80.66 80.66 AIR TEN V.EFF 81.15	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 16615 1600 1600 1600 1600 1600 1600 1600	1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466 1.466 1.475	FUEL 0.046 0.045 0.045 0.053 0.053 0.053 0.060 0.060 0.060 FUEL 0.046	000 RPM A/F 32.599 32.542 32.665 32.379 27.904 27.863 27.879 24.4522 24.522 24.574 24.521 24.521 24.522 24.742	PERCONSTRUCTION OF THE PROPERTY OF THE PROPERT	ENT WAT BMEP 42.08 42.08 42.08 42.60 52.60 52.60 53.11 63.11 WAT BMEP 42.08	FER = 15 BSFC 0.4004 0.4015 0.5992 0.4045 0.5573 0.5578 0.5583 0.5264 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.0 3	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.65 81.15 81.15 80.66 80.66 AIR TEN V.EFF 81.15 81.15	E.TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 16615 1600 1600 1600 1600 1600 1600 1600	1.484 1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466 1.466 1.475 1.475	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.050 0.060 0.060 FUEL 0.046 0.046	32.599 32.549 32.549 32.645 32.645 32.79 27.928 27.904 27.863 27.863 24.474 24.522 24.474 24.522 24.474 24.522	PERCONSTRUCTION OF THE PROPERTY OF THE PROPERT	ENT WAT AT A	ER = 15 BSFC 0.6004 0.6015 0.5578 0.5578 0.5588 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 32.0 3	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 80.66 80.66 80.66 AIR TEN V.EFF 81.15 91.15	E.TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 6F.= 88 F E.TEMP 3447.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 166150000044444	1.484 1.484 1.475 1.475 1.475 1.475 1.466 1.466 1.466 1.475 1.475 1.475	FUEL 0.046 0.050 0.060 0.060 0.060 0.060 0.060 0.0646 0.046 0.046 0.046 0.046 0.046	32.599 32.549 32.549 32.645 32.645 32.645 32.4904 27.863 27.863 27.882 24.474 24.522 24.474 24.522 24.522 24.522	PERCONSTRUCTION OF CONTRACT OF	MAT WAT 108 42.08 42.600 52.6000 52.600 52.600 52.600 52.600 52.600 52.600 52.600 52.600 52.6	FER = 15 BSFC 0.4004 0.4015 0.5992 0.4045 0.5578 0.5586 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 32.0 32.0 32.0 32.0 32.0 32.0 32.0 32.0 32.0 32.0 32.0	O.00 O.00 O.00 O.00 O.00 O.00 O.00 O.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 81.15 80.66 80.66 80.66 80.66 80.66 80.66 80.66 80.66	E.TEMP 356.00 356.00 356.00 405.00 405.00 4062.00 462.00 462.00 462.00 518. = 88 F	% GP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 164512000000000000000000000000000000000000	1.4844 1.4844 1.4755 1.4755 1.4755 1.46666 1.46 1.4755 1.4755 1.4755 1.4755 1.4755 1.4755	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.060 0.060 0.060 0.060 0.060 0.060 0.060	A/F 32.599 32.542 32.665 32.379 27.928 27.904 27.863 27.879 24.452 24.522 24.522 24.522 24.522 24.522 27.863 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879	PERCONSTRUCTION OF THE PROPERTY OF THE PROPERT	ENT WAT AT A	ER = 15 BSFC 0.4004 0.4015 0.5992 0.4057 0.5578 0.5586 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 32.0 3	NTAKE ID 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.64 81.15 81.15 80.66 80.66 80.66 AIR TEN V.EFF 81.15 91.15	E.TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 56.2 E.TEMP 345.0 345.0 371.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
EN D 666500000044444 = FG D 6665000000000000000000000000000000000	1.484 1.484 1.475 1.475 1.475 1.4666 1.4666 1.4675 1.4775 1.4775 1.406	FUEL 0.046 0.053 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.065 0.065 0.055	A/F 32.599 32.542 32.542 32.665 32.329 27.904 27.863 27.879 24.422 24.522 24.522 24.522 24.148 32.148 32.148 32.148 32.179 27.590 27.720	PER P 99999777744444 ORC P B 55554888 P B 5449777744444 ORC P B 55554888 P B 5555488 P B 555548 P B 5555	WAT BM EP 88 422.040 522.450 1133 WAT BM EP 088 422.050 532.133 WAT BM EP 088 422.050 500 500 500 500 500 500 500 500 500	ER = 15 BSFC 0.4004 0.4015 0.5578 0.5578 0.5586 0.55258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.0 3	O.00 O.00 O.00 O.00 O.00 O.00 O.00 O.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 81.15 80.66 80.66 80.66 80.66 80.66 80.66 80.66 80.66	E. TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 5F. = 88 F E. TEMP 347.0 347.0 3791.0	% GP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG LD 164512000000000000000000000000000000000000	1.484 1.484 1.475 1.475 1.475 1.475 1.4666 1.4666 1.475 1.475 1.475 1.475 1.475 1.4666	FUEL 0.046 0.045 0.045 0.050 0.050 0.050 0.060 0.060 0.060 0.060 0.060 0.060 0.060	A/F 32.599 32.542 32.665 32.379 27.928 27.904 27.863 27.879 24.452 24.522 24.522 24.522 24.522 24.522 27.863 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879 27.879	PERCONSTRUCTION OF THE PROPERTY OF THE PROPERT	MAT WAT NO B A 12.08	ER = 15 BSFC 0.4004 0.4015 0.5992 0.4057 0.5578 0.5586 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 32.0 3	NTAKE ID 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 81.15 80.66 80.66 AIR TEN V.EFF 81.15 81.15 81.15 81.15 81.15	E.TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 56.2 E.TEMP 345.0 345.0 371.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
EN D 666500000044444 = FG D 6665000000000000000000000000000000000	1.484 1.484 1.475 1.475 1.475 1.4666 1.4666 1.4675 1.4775 1.4775 1.406	FUEL 0.046 0.053 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.060 0.065 0.065 0.055	A/F 32.599 32.542 32.542 32.665 32.329 27.904 27.863 27.879 24.422 24.522 24.522 24.522 24.148 32.148 32.148 32.148 32.179 27.590 27.720	PER P 99999777744444 ORC P B 55554888 P B 5449777744444 ORC P B 55554888 P B 5555488 P B 555548 P B 5555	WAT BM EP 88 422.040 522.450 1133 WAT BM EP 088 422.050 532.133 WAT BM EP 088 422.050 500 500 500 500 500 500 500 500 500	ER = 15 BSFC 0.4004 0.4015 0.5578 0.5578 0.5586 0.55258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.0 3	NTAKE ID 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 81.15 80.66 80.66 AIR TEN V.EFF 81.15 81.15 81.15 81.15 81.66 80.66	E. TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 5F. = 88 F E. TEMP 347.0 347.0 3791.0	% OP 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG L 64444 ENG L 644444 ENG L 64444 ENG L 64	1.484 1.484 1.475 1.475 1.475 1.475 1.4666 1.4666 1.475 1.475 1.475 1.475 1.475 1.4666	FUEL 0.046 0.050 0.060 0	000 RPM A/F 32.599 32.542 32.542 32.579 27.904 27.863 27.879 24.452 24.522 24.522 24.522 24.108 32.148 32.108 32.148 32.109 27.750	PER P 999 4.54497777774444 000 PE H 4.5648888 PE H 44.54497777744444 000 PE H 44.544977774444 000 PE H 44.5449777774444 000 PE H 44.544855555555555555555555555555555555	WAT 08 422.080 422.0400 552.451117 WAT 88 422.080 552.451117 WAT 88 422.080 600 600 600 600 600 600 600 600 600	ER = 15 BSFC 0.4004 0.4015 0.5992 0.45573 0.5578 0.5583 0.5248 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.0 3	O.00 O.00 O.00 O.00 O.00 O.00 O.00 O.00	AIR TEN V.EFF 81.64 81.64 81.15 81.15 80.66 80.66 AIR TEN V.EFF 81.15 81.15 81.15 81.66 80.66 80.66	E. TEMP 356.0 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 56.0 E. TEMP 347.0 547.0 5791.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
ENG L 66650000044444	1.4844 1.4884 1.4775 1.44666 1.446666 1.4466666 1.44775 1.44775 1.44775 1.44775 1.4466666666666666666666666666666666666	FUEL 0.046 0.050 0	000 RPM A/F 32.599 32.5425 32.579 27.879 27.8879 27.8879 24.4522 24.522 24.522 24.522 24.522 24.742 24.522 24.752 27.750 27.750 27.630	PER P 999997777744444 OCC 44.566888 P 9999977777744444 OCC 5.55566888 P 8 555449977777	WAT 100 B 472.008 472.600 572.1177 WAT 1008 472.008 47	FER = 15 BSFC 0.4004 0.4015 0.5578 0.5578 0.5578 0.5588 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.0 31.5 31.5 31.5 32.0 3	O.00 O.00 O.00 O.00 O.00 O.00 O.00 O.00	AIR TEN V.EFF 81.644 81.653 81.155 80.6666 80.6666 AIR TEN 91.155 81.155 81.155 81.155 81.155 81.666666	E. TEMP 356.0 356.0 356.0 405.0 405.0 405.0 462.0 462.0 462.0 591.0 591.0	% OF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	
EN D 66640000044444	1.4844 1.4844 1.4755 1.4775 1.4755 1.44666 1.4666 1.4666 1.4755 1.4775 1.4766666 1.46666666666666666666666666666666	FUEL 0.046 0.050 0.060 0	32.599 32.542 32.542 32.645 32.645 32.6479 27.904 27.863 27.863 27.4482 24.474 24.52 24.474 24.52 24.52 24.52 27.630 27.630 27.630 27.630 24.653	98 P 999997777744444 OR P 999997777444444 OR P 999997777444445555555555555555555555555	WAR P 088 472.008 472.	FER = 15 BSFC 0.4004 0.4015 0.5578 0.5578 0.5578 0.5588 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258 0.5258	30.0 INJ 31.0 31.0 31.0 31.5 31.5 31.5 32.0 3	NTAKE ID 0.00 0.00 0.00 0.00 0.00 0.00 0.00	AIR TEN V.EFF 81.644 81.658 81.158 81.158 80.666 80.666 AIR TEN 81.158 81.158 80.6666 80.6666 80.6666 80.6666	#P.= 88 F E.TEMP 356.00 3556.00 3556.00 405.00 405.00 405.00 462.00 462.00 FP.= 88 F E.T EMP 3447.00 3791.00 3791.00 3791.00 3791.00 3791.00	% GF 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.	

TABLE IV

COMPUTER PRINTOUT OF THE CYLINDER PRESSURE
AND NEEDLE LIFT DIGITIZED DATA POINTS

	DATA CORRESPONDS	TO 71724IM.ASC	**************
I NL(I)	S%L(I)	CP(I)	SCP(I)
I NL(I) 192	SXL(I) -1 00 10 -1 00 00 10 00 11 00 00 11 00 00 11 157 177 -37 -38 -37 -38 -10 -3	I 2331852087666780360507432344445555666677778888889996 CP 2331852087666780360507432344445555666677778888889996	SCP(I) 6 5 6 7 7 7 8 8 9 9 10 10 11 11 12 13 13 14 15 17 17 19 19 21 22 24 24 26 26 28 29 31 31 30 31 31 31 29 29 26 33 21 20 30 20 30 21 20 30
239 1582 240 1600	18 -10	906 928	22 32



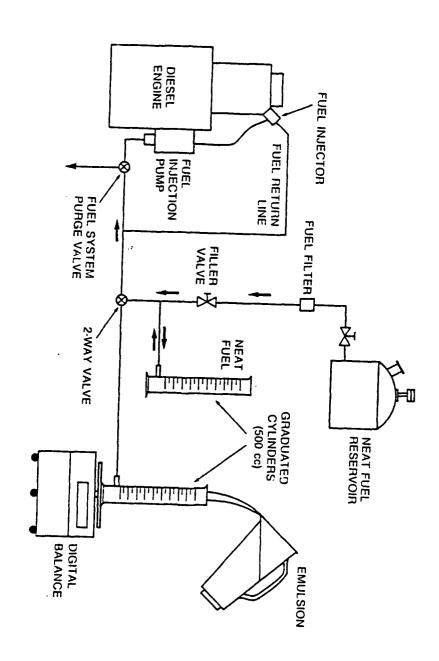


FIGURE 2 SCHEMATIC OF THE FUEL DELIVERY SYSTEM

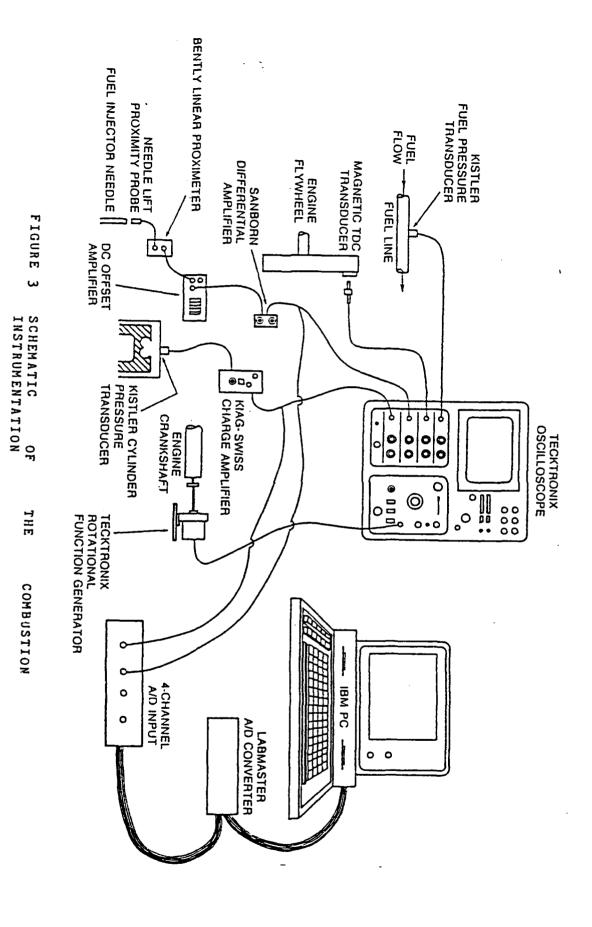
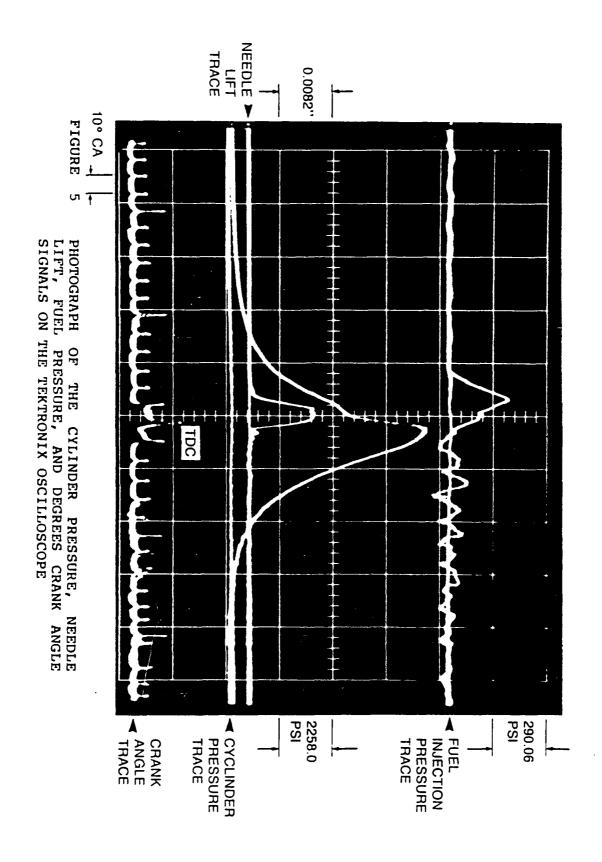
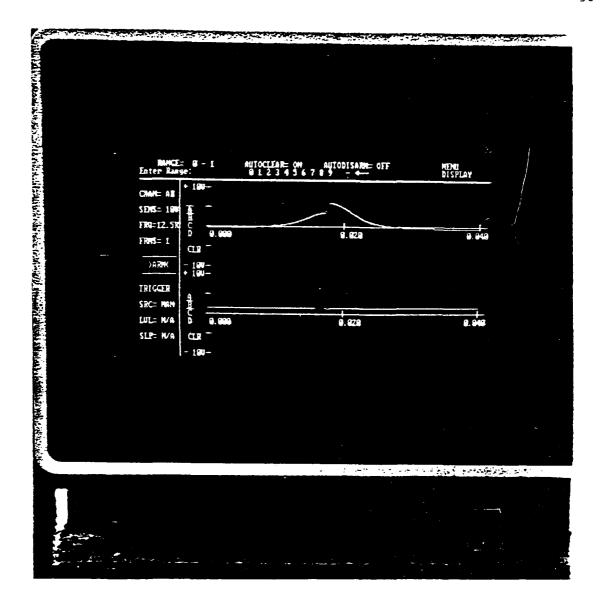




FIGURE 4 PHOTOGRAPH OF THE COMBUSTION DIAGNOSTIC INSTRUMENTATION USED TO DETERMINE THE IGNITION DELAY PERIOD





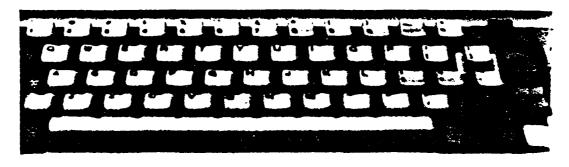


FIGURE 6 PHOTOGRAPH OF THE NEEDLE LIFT AND CYLINDER PRESSURE DIGITIZED SIGNALS ON THE COMPUTER SCREEN

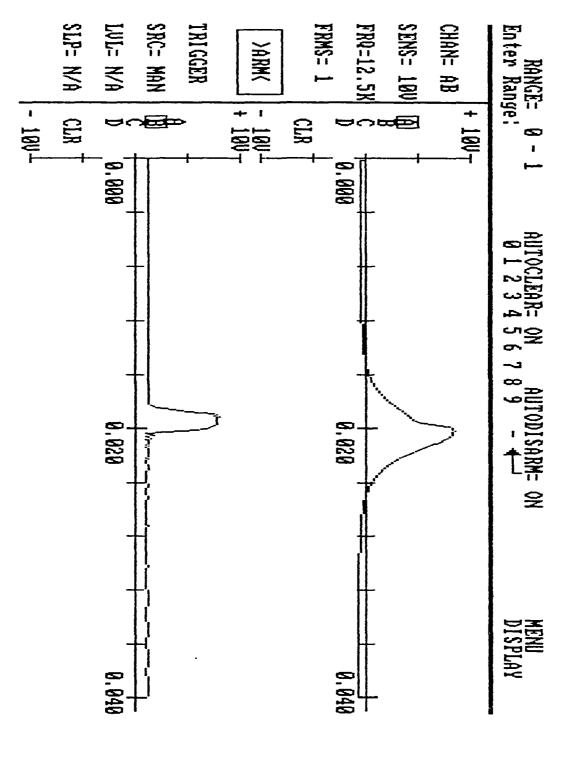


FIGURE 7 IBM PC COMPUTER PLOT OF THE CYLINDER PRESSURE AND NEEDLE LIFT SIGNALS

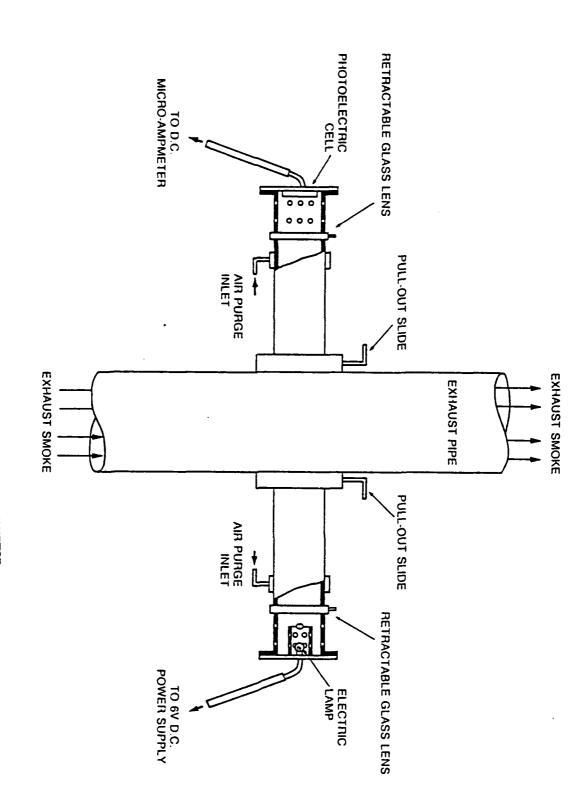


FIGURE 8 SCHEMATIC OF THE SMOKE MONITOR

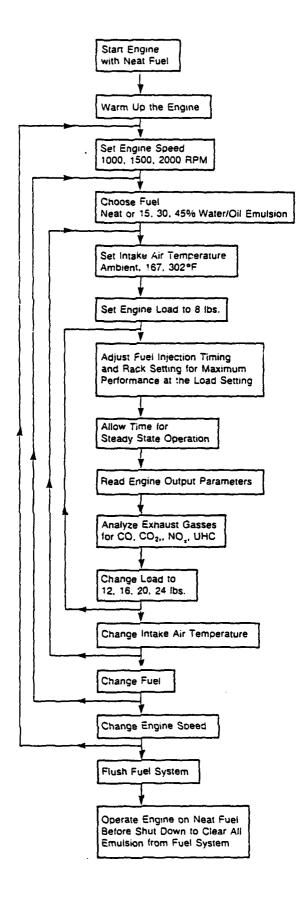


FIGURE 9 FLOW CHART OF TESTING PROGRAM

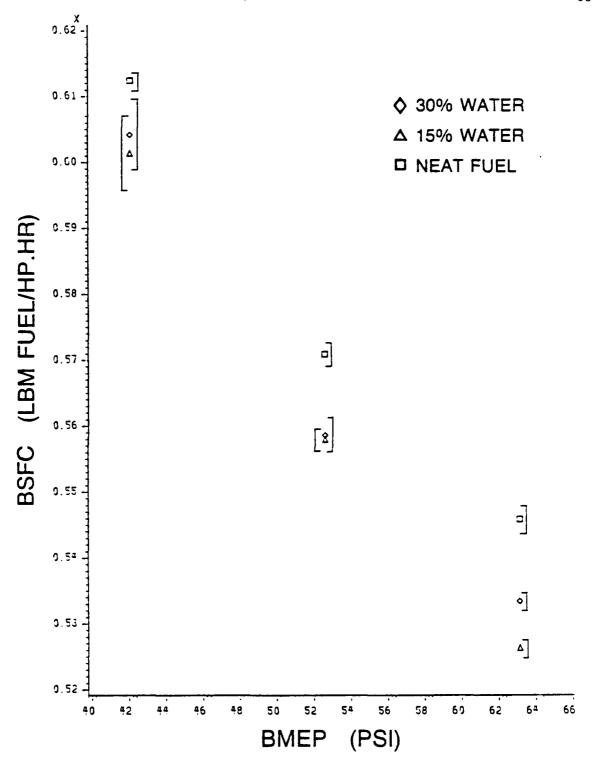


FIGURE 10 EFFECT OF WATER / OIL EMULSIONS ON THE BSFC FOR AN INTAKE AIR TEMPERATURE OF 88.0 F AT 2000 RPM

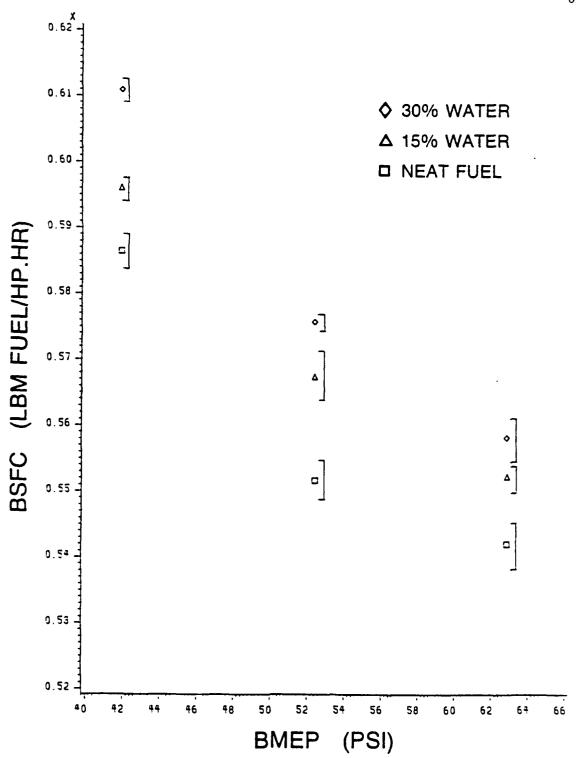


FIGURE 11 EFFECT OF WATER / OIL EMULSIONS ON THE BSFC FOR AN INTAKE AIR TEMPERATURE OF 88.0 F AT 1000 RPM

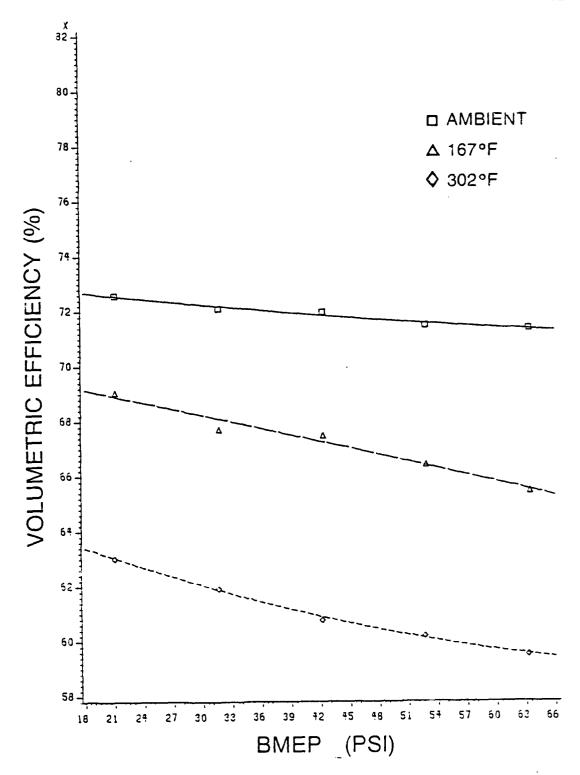


FIGURE 12 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE VOLUMETRIC EFFICIENCY FOR NEAT FUEL AT 1000 RPM

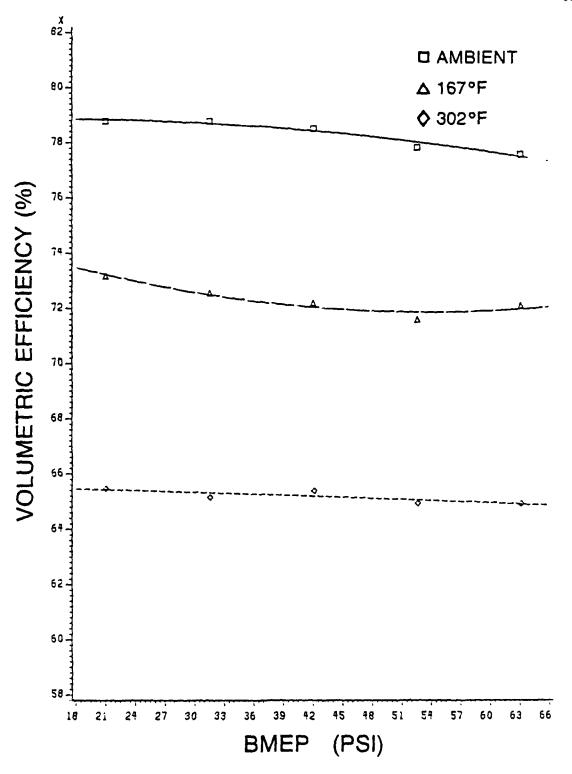


FIGURE 13 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE VOLUMETRIC EFFICIENCY FOR NEAT FUEL AT 2000 RPM

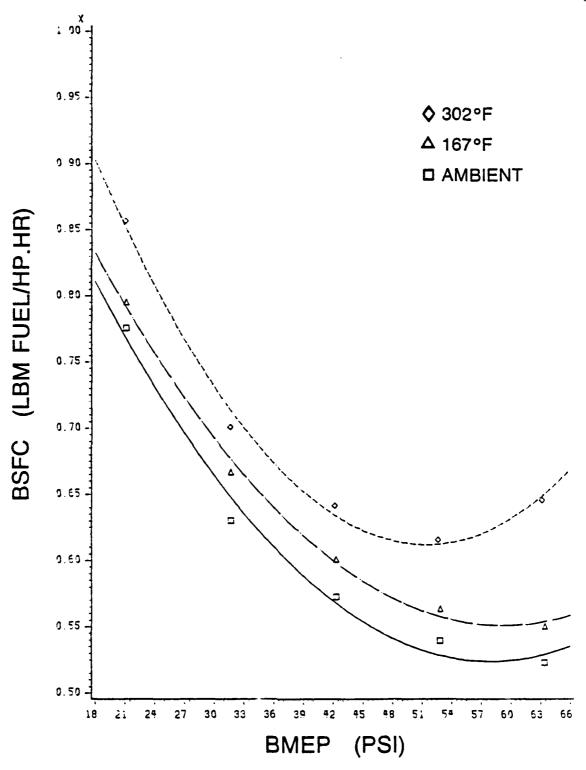


FIGURE 14 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR NEAT FUEL AT 1000 RPM

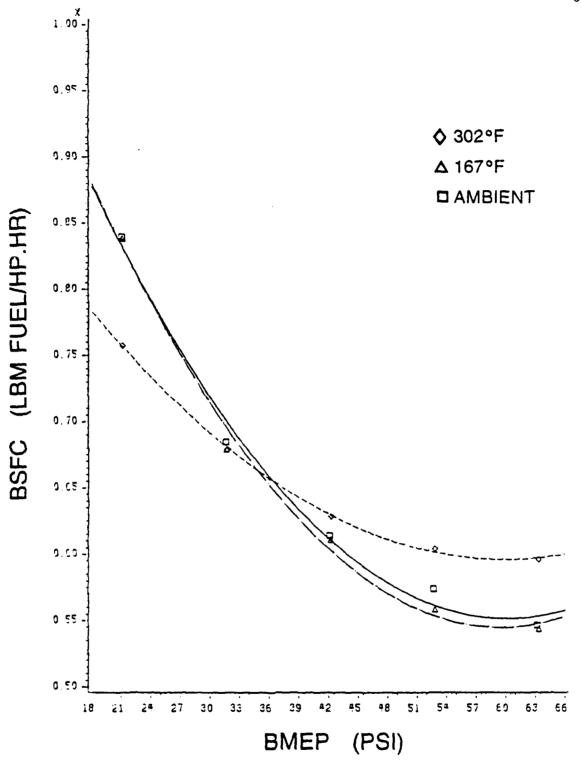


FIGURE 15 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 15 % WATER / OIL EMULSION AT 1000 RPM

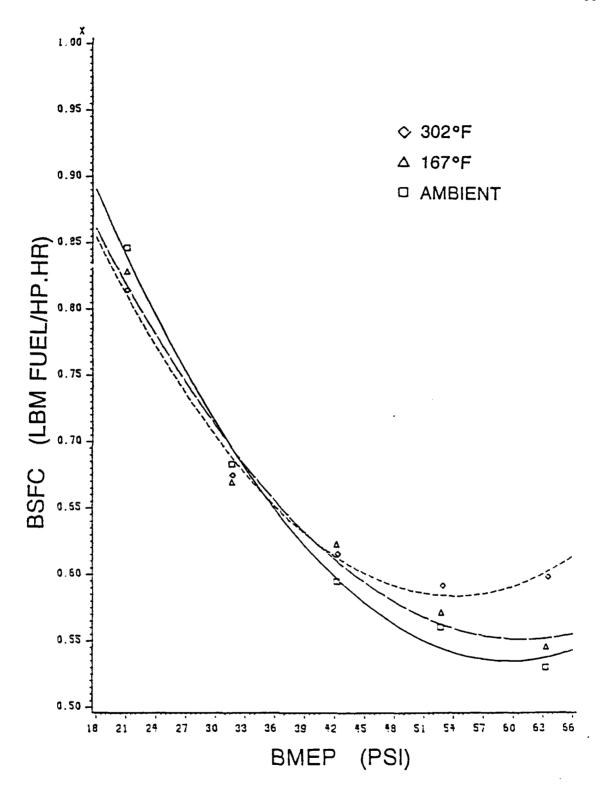


FIGURE 16 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 30 % WATER / OIL EMULSION AT 1000 RPM

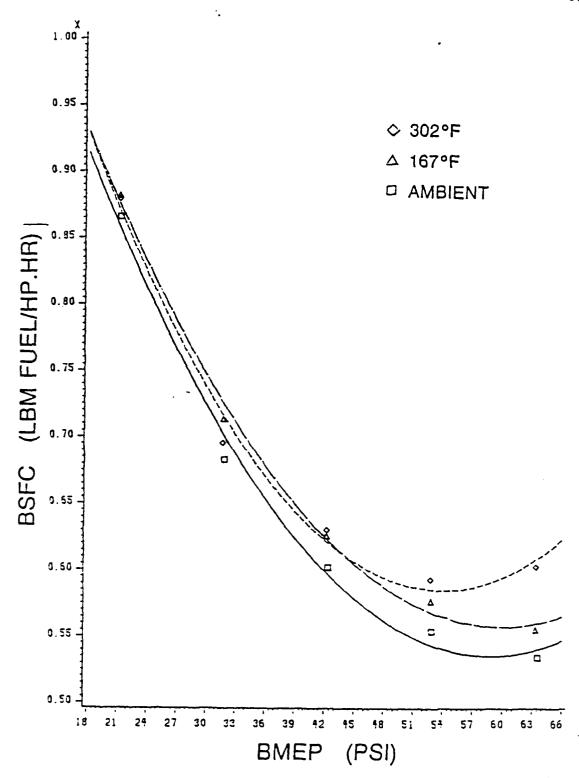


FIGURE 17 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 45 % WATER / OIL EMULSION AT 1000 RPM

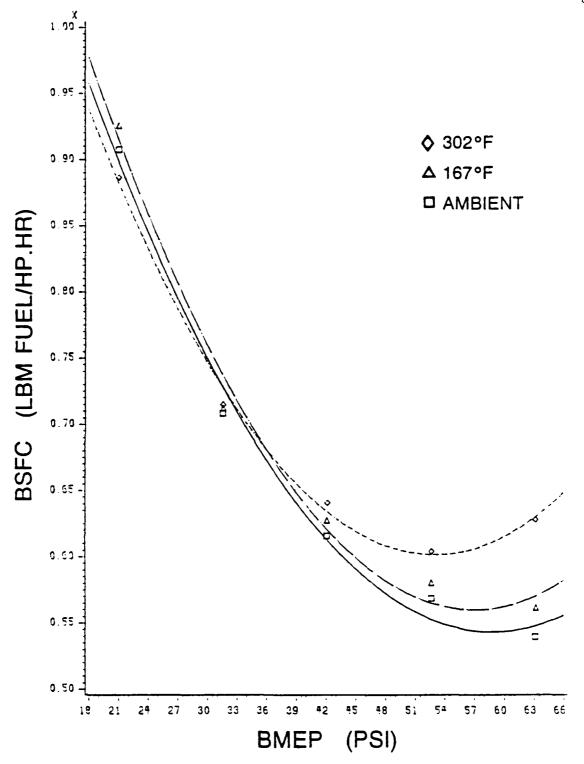


FIGURE 18 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR NEAT FUEL AT 2000 RPM

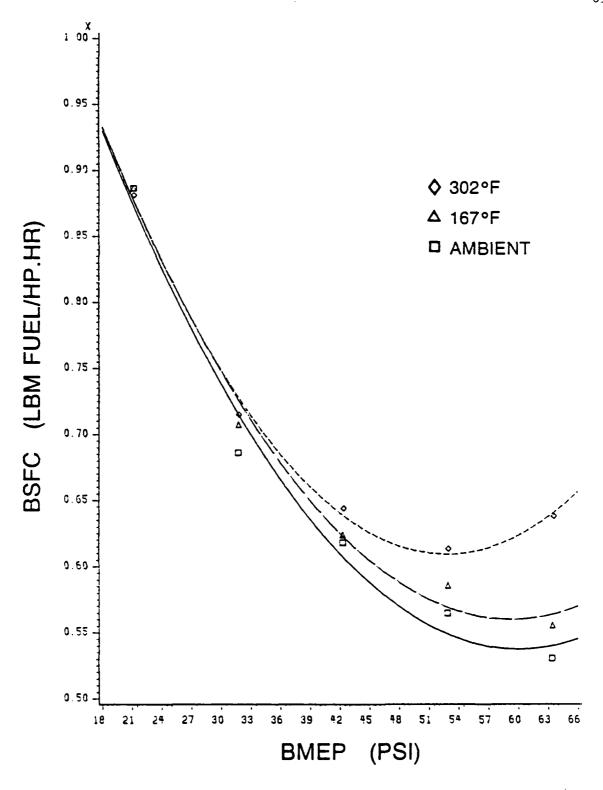


FIGURE 19 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 15 % WATER / OIL EMULSION AT 2000 RPM

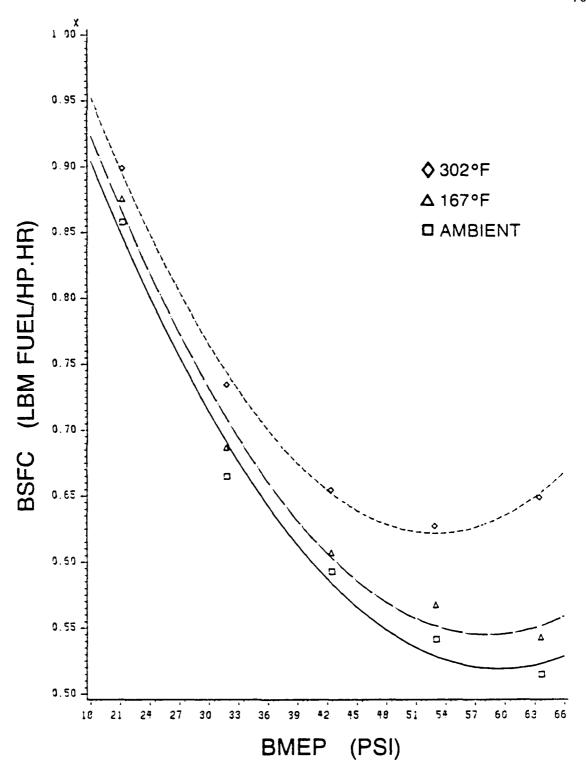


FIGURE 20 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 30 % WATER / OIL EMULSION AT 2000 RPM

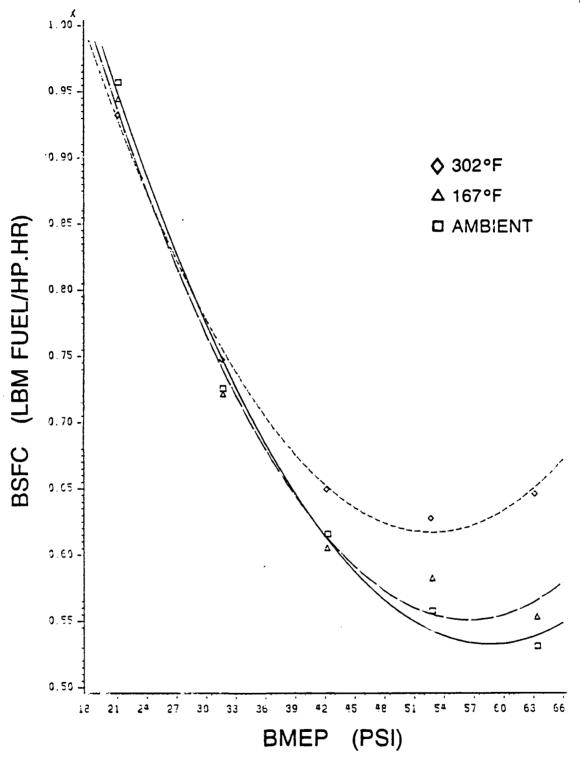


FIGURE 21 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE BSFC FOR 45 % WATER / OIL EMULSION AT 2000 RPM

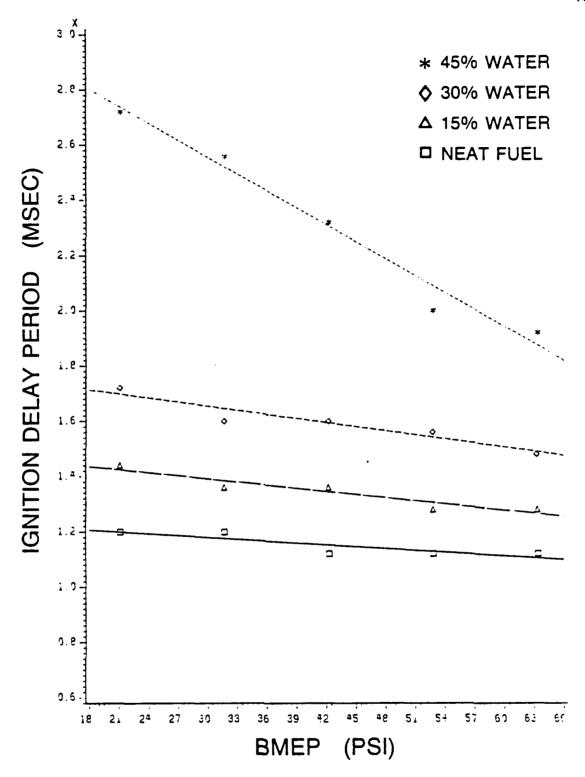


FIGURE 22 EFFECT OF WATER / OIL EMULSIONS ON THE IGNITION DELAY PERIOD FOR AN INTAKE AIR TEMPERATURE OF 167 F AT 1000 RPM

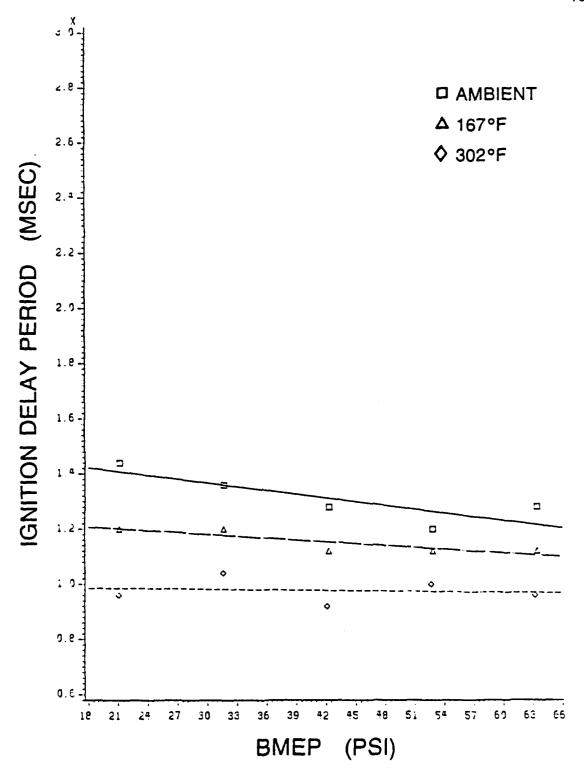


FIGURE 23 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE IGNITION DELAY PERIOD FOR NEAT FUEL AT 1000 RPM

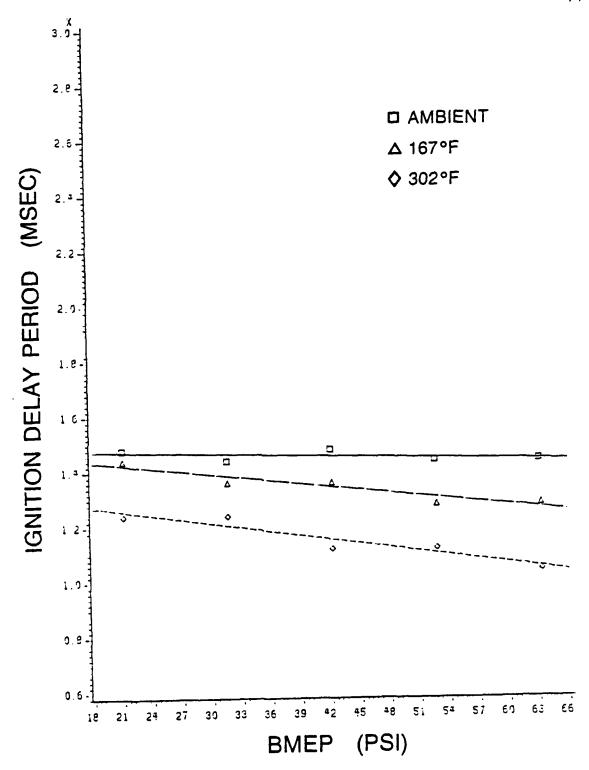


FIGURE 24 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE IGNITION DELAY PERIOD FOR 15 % WATER / OIL EMULSION AT 1000 RPM

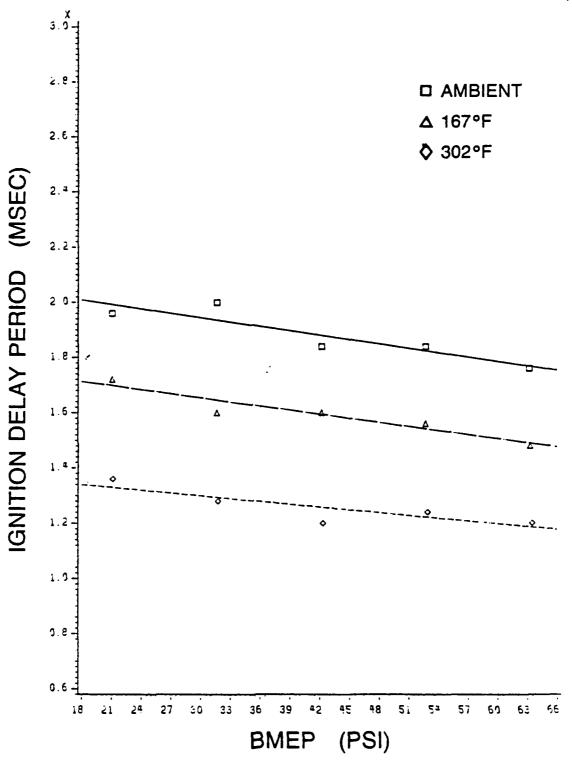


FIGURE 25 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE IGNITION DELAY PERIOD FOR 30 % WATER / OIL EMULSION AT 1000 RPM

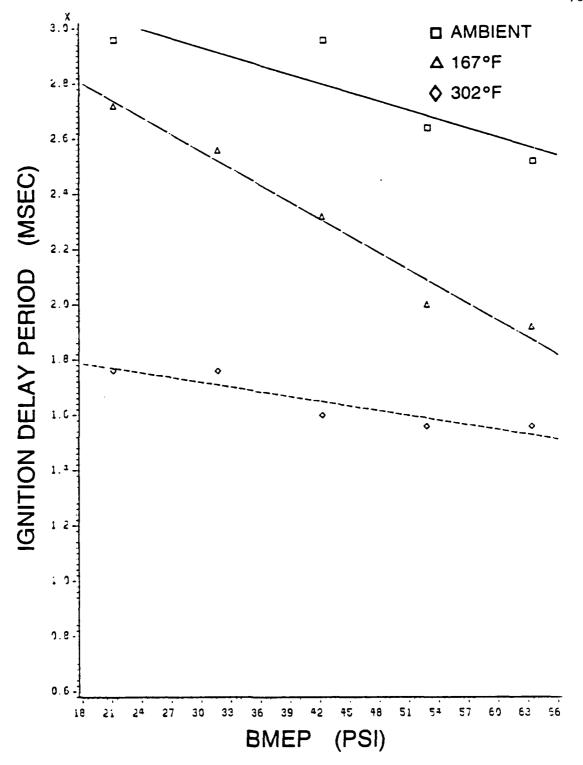


FIGURE 26 EFFECT OF ENGINE LOAD AND INTAKE AIR TEMPERATURE ON THE IGNITION DELAY PERIOD FOR 45 % WATER / OIL EMULSION AT 1000 RPM

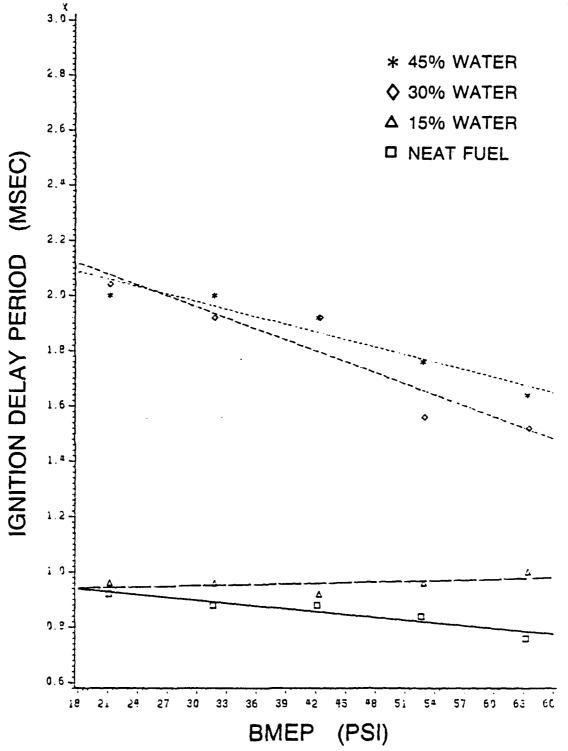


FIGURE 27 EFFECT OF WATER / OIL EMULSIONS ON THE IGNITION DELAY PERIOD FOR AN INTAKE AIR TEMPERATURE OF 167 F AT 2000 RPM

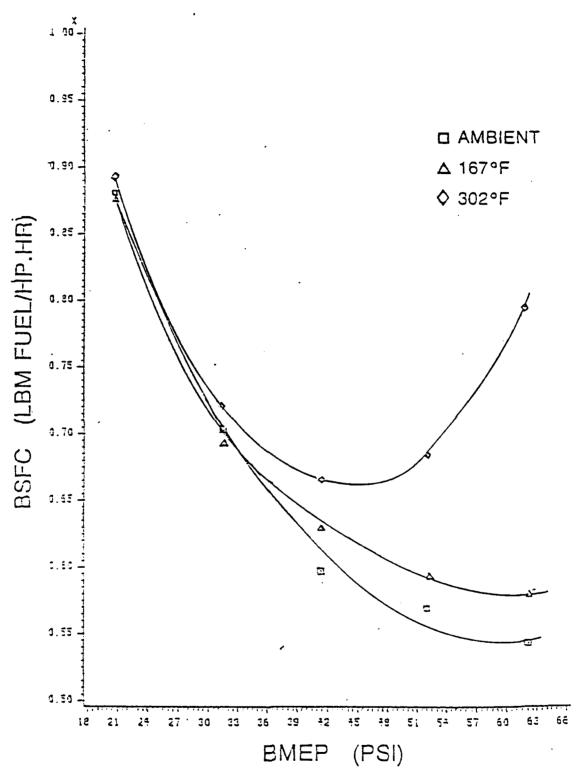


Figure 28 Effect of Engine Load and Intake Air Temperature on the BSFC for Neat JP-4 Fuel at 2000 RPM.

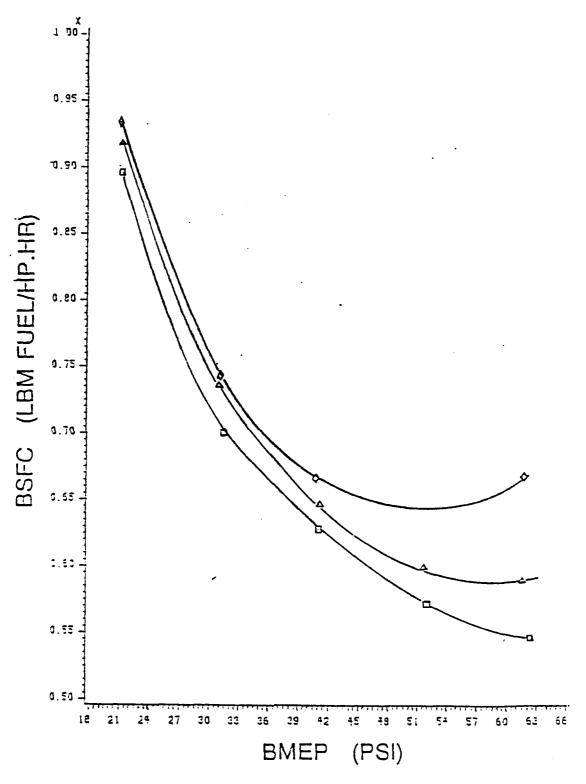


Figure 29 Effect of Engine Load and Intake Air Temperature on the BSFC for 15% w/JP-4 Emulsion at 2000 RPM

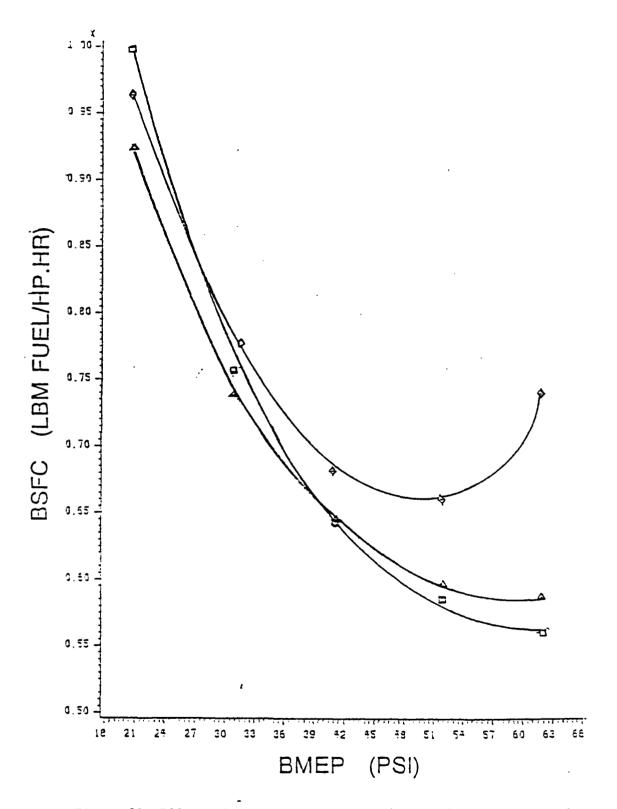


Figure 30 Effect of Engine Load and Intake Air Temperature on the BSFC for 30% $\mbox{w/JP-4}$ Emulsion at 2000 RPM

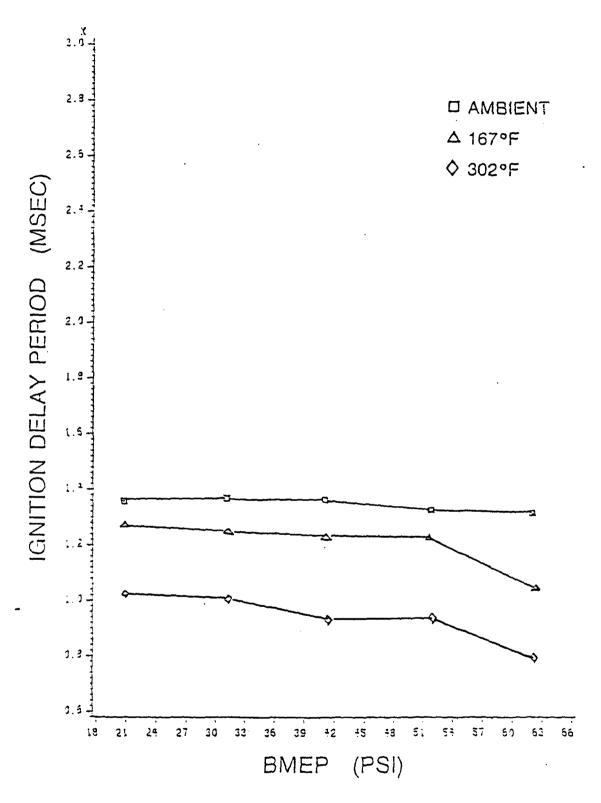


Figure 31 Effect of Engine Load and Intake Air Temperature on the Ignition Delay for Neat JP-4 Fuel at 2000 RPM

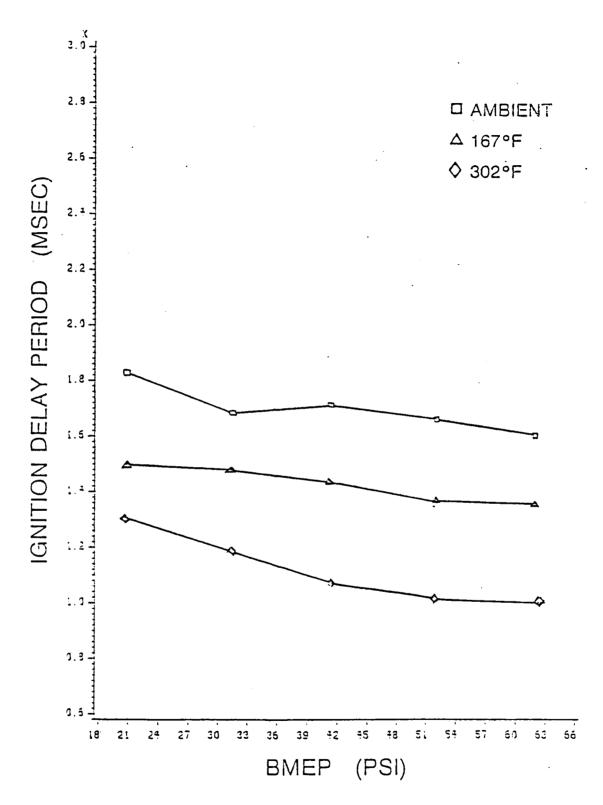


Figure 32 Effect of Engine Load and Intake Air Temperature on the Ignition Delay Period for 15% w/JP-4 Emulsion at 2000 RPM

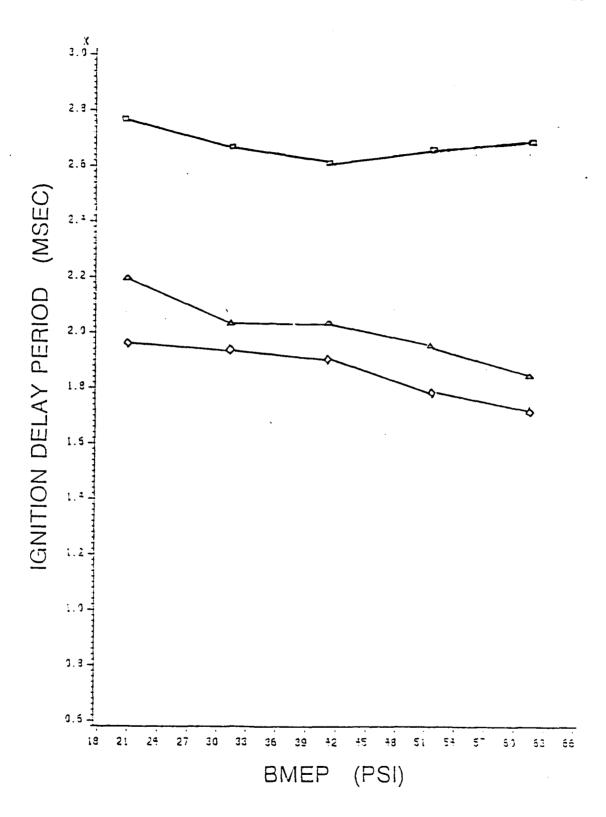


Figure 33 Effect of Engine Load and Intake Air Temperature on the Ignition Delay Period for 30% $\rm w/JP-4$ Emulsion at 2000 RPM

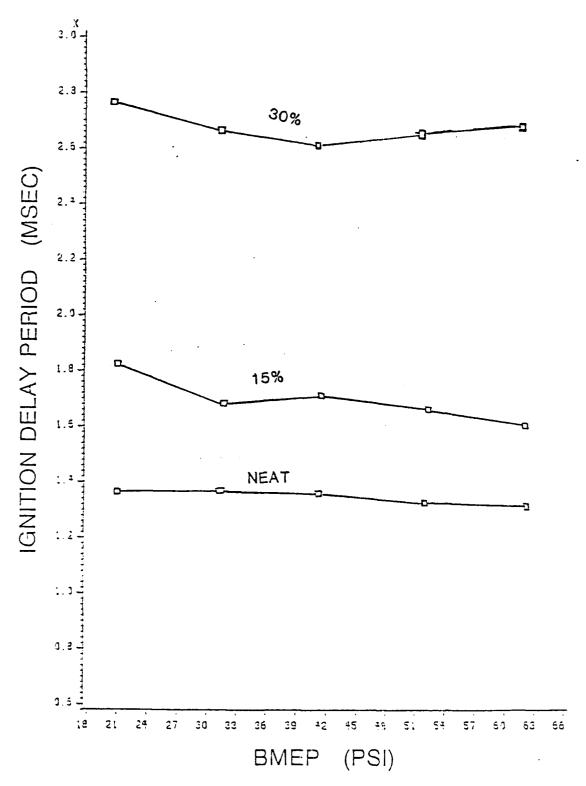
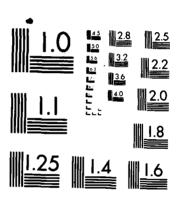


Figure 34 Effect of Water /JP-4 Emulsions on the Ignition Delay Period for Ambient Intake Air Temperature at 2000 RPM





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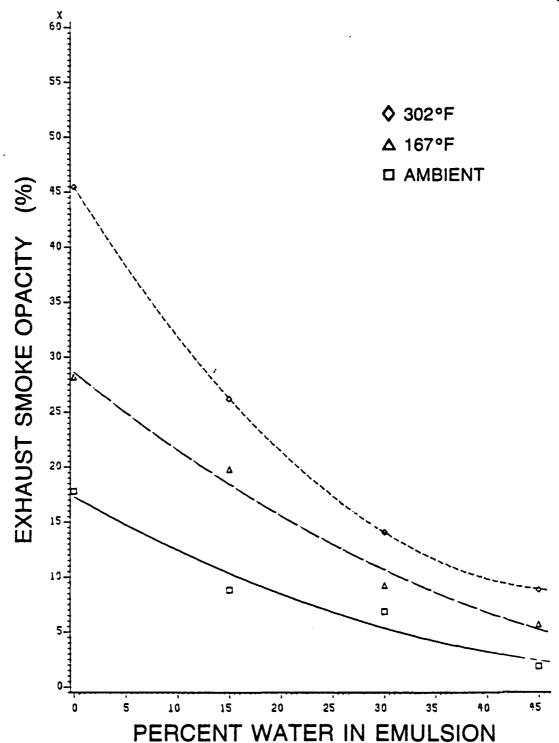


FIGURE 35 EFFECT OF WATER / OIL EMULSIONS AND INTAKE AIR TEMPERATURE ON THE EXHAUST GAS OPACITY FOR BMEP=63 PSI AT 1000 RPM

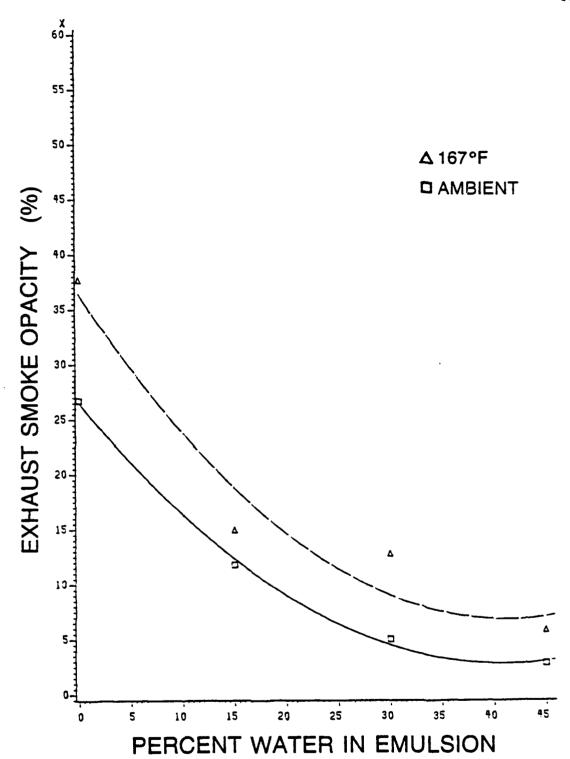


FIGURE 36 EFFECT OF WATER / OIL EMULSIONS AND INTAKE AIR TEMPERATURE ON THE EXHAUST GAS OPACITY FOR BMEP=63 PSI AT 2000 RPM

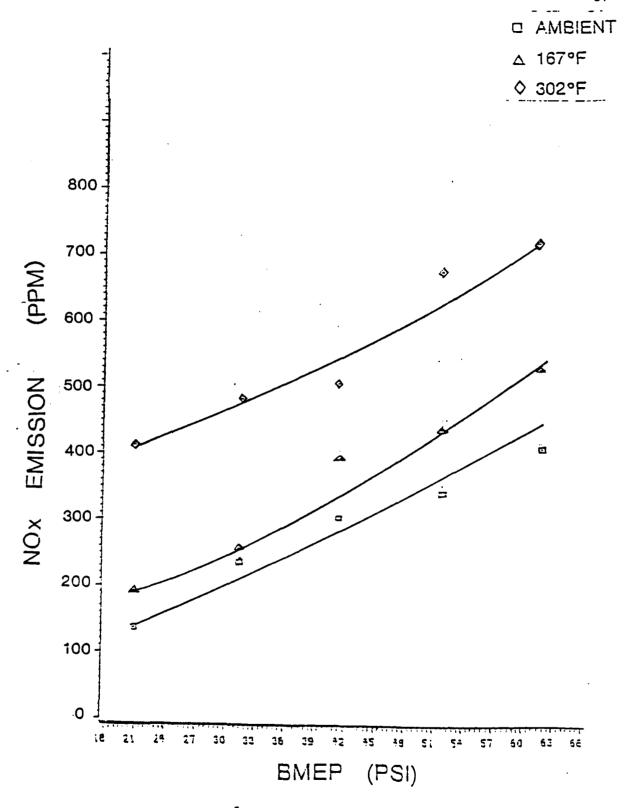


Figure 37 Effect of Engine Load and Intake Air Temperature on Nox Emission for Neat Diesel Fuel #2 at 2000 RPM

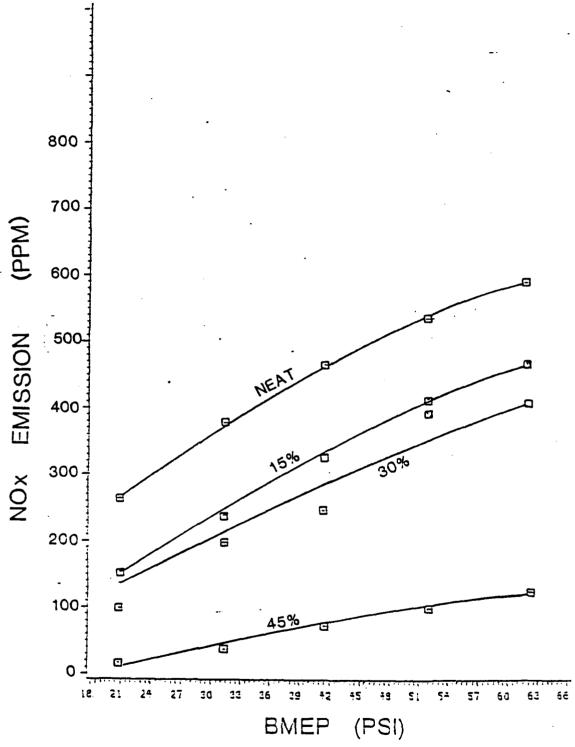


Figure 38 Effect of Engine Load and Water/Diesel Emulsion on Nox Emission at Ambient Intake Air Temperature and 2000 RPM

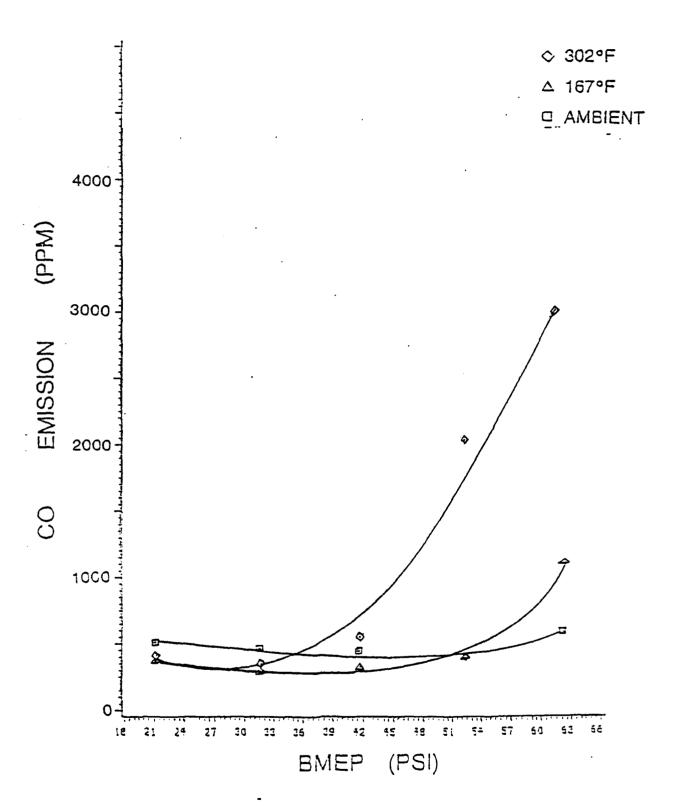


Figure 39 Effect of Engine Load and Intake Air Temperature on Co Emission for Neat Diesel Fuel #2 at 2000 RPM

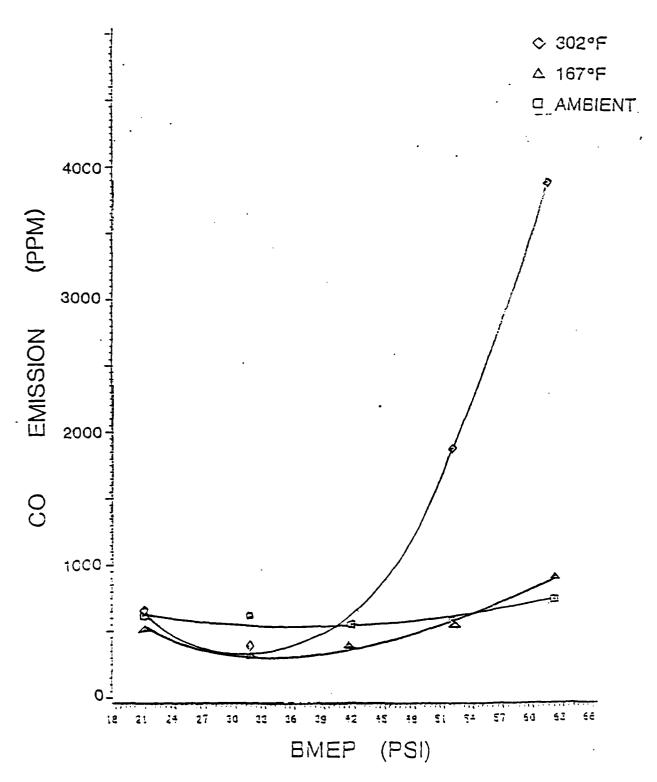


Figure 40 Effect of Engine Load and Air Intake Temperature on Co Emission for w/Diesel Emulsion at 2000 RPM

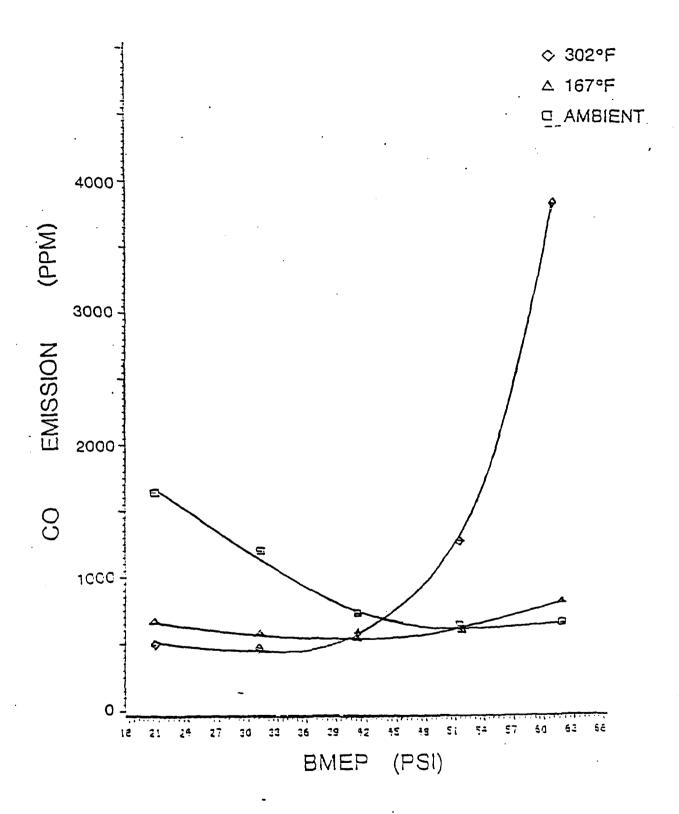


Figure 41 Effect of Engine Load and Air Intake Temperature on Co Emission For 30% w/Diesel Emulsion at 2000 RPM

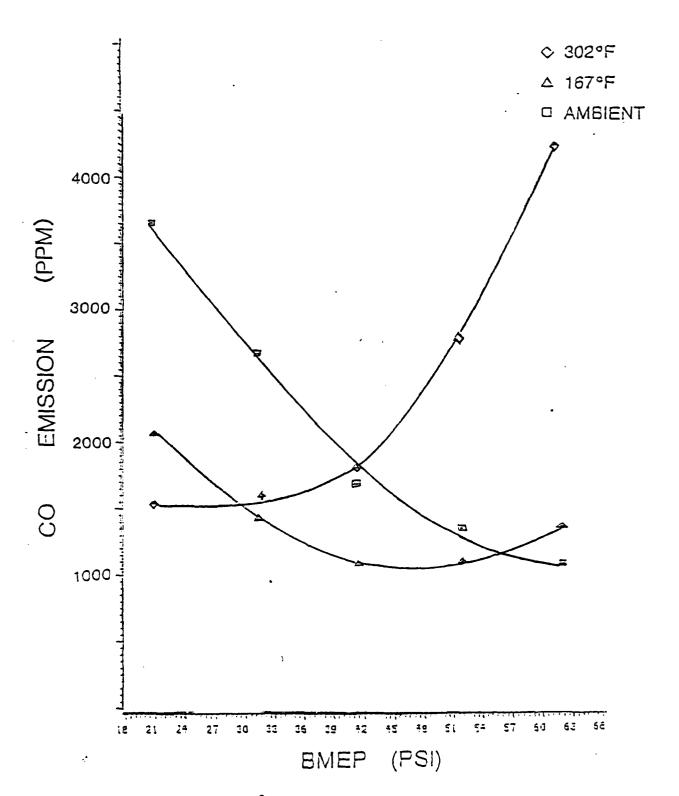


Figure 42 Effect of Engine Load and Intake Air Temperature on Co Emission for 45% w/Diesel Emulsion at 2000 RPM

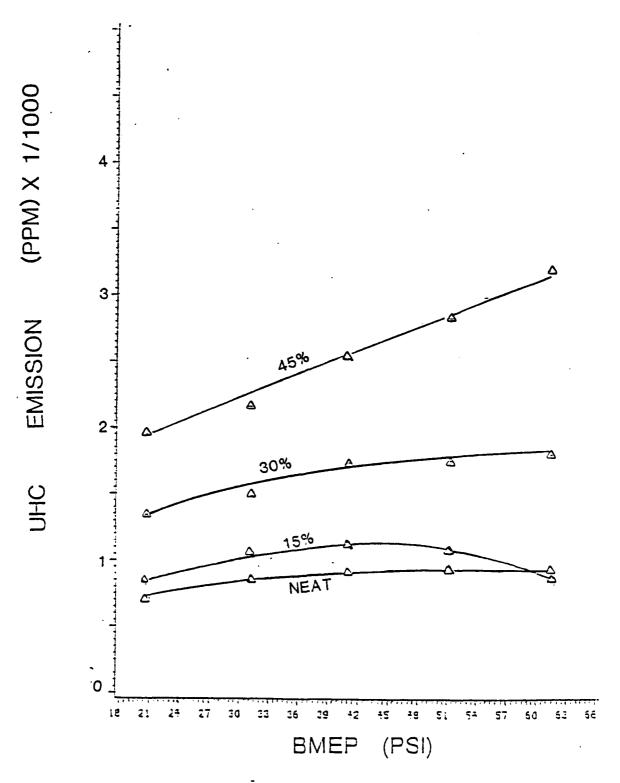


Figure 43 Effect of Engine Load and Water/Diesel Emulsion at $167\,^{\circ}\text{F}$ Intake Air Temperature and 2000 RPM

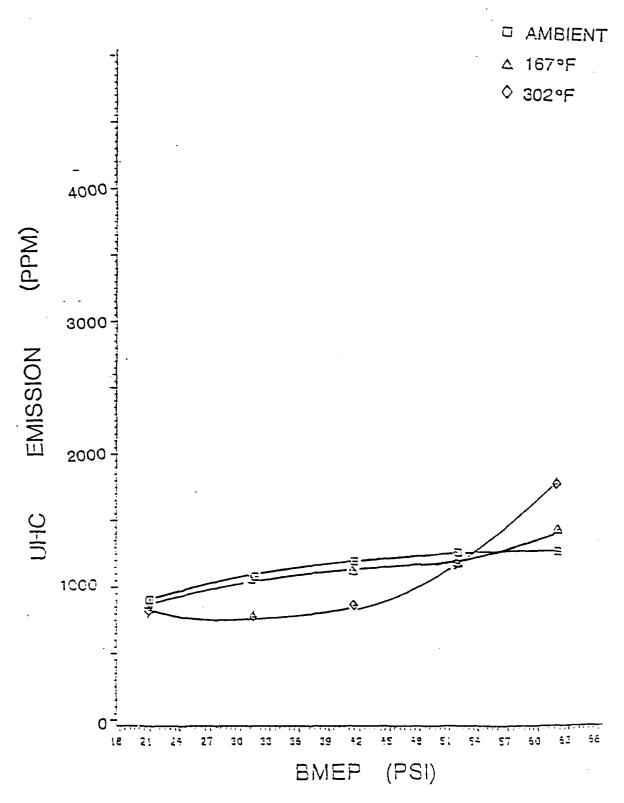


Figure 44 Effect of Engine Load and Intake Air Temperature on UHC Emission for Diesel Fuel #2 at 1000 RPM

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